# Data-Driven Modeling of the Magnetic Profile and Rotation Profile for Advanced Tokamak Scenarios in DIII-D

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#### 53<sup>rd</sup> Annual Meeting of the APS Division of Plasma Physics

This work was supported by the NSF CAREER Award Program ECCS-0645086 and the US Department of Energy (DE-FG02-09ER55064, DE-FC02-04ER54698, and DE-FG02-08ER85195)



November 17, 2011



First-principle predictive models based on flux averaged transport equations often yield complex expressions not suitable for real-time control. As an alternative to first-principle modeling, data-driven modeling techniques involving system identification have the potential to obtain low-complexity, dynamic models without the need for ad hoc assumptions. This work focuses on the evolution of the toroidal rotation and safety factor profiles in response to magnetic, heating and current-drive systems. Experiments are conducted during the current flattop, in which the actuators are modulated in open-loop to obtain data for the model identification. The plasma profiles are discretized in the spatial coordinate by Galerkin projection. Then a linear model is generated by the prediction error method to relate the rotation and safety factor profiles to the actuators according to a least squares fit.

- Simple linear models based on system identification (data-driven modelling) are desired for control implementations
- Models of magnetic profile and toroidal rotation profile in response to certain inputs; the neutral beams, the gyrotron power, and the plasma current.
- A model for coupled evolution of the magnetic profile and rotation profile

- Achieving sustained tokamak operation.
- Non-inductive sources of current are required for steady state operation.
- Setting up a suitable toroidal current profile can lead to self-generated, non inductive current (bootstrap current).
- Controlling the current profile will therefore be important to achieving steady-state reactor operation.

# Why rotation profile control?

- Plasma performance while operating in high pressure conditions is limited by transport phenomena and Magnetohydrodynamic (MHD) instabilities.
- Optimizing some of the plasma profiles such as the toroidal rotation profile can improve plasma performance.
- For example, increasing bulk fluid rotation around the tokamak produces a velocity gradient. The velocity results in a sheared plasma flow reducing turbulence and improving heat confinement.

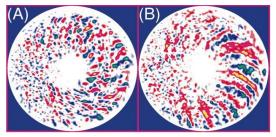


Figure: Simulation of turbulence with (A) and without (B) sheared plasma flow. [1] M. De Bock, *Understanding and controlling plasma rotation in tokamaks*, Doctoral Thesis, Technische Universiteit Eindhoven, 2007

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- First-principle models based on transport equations yield complex expressions not suitable for control.
- Use linear models based on system identification instead.
- Around certain trajectories the PDEs can be linearized as

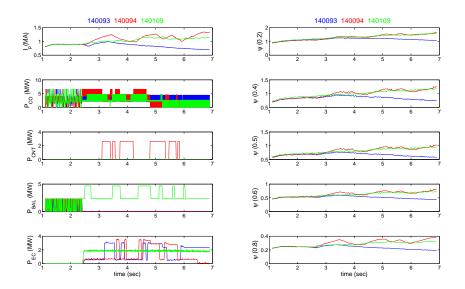
$$\frac{\partial x(\hat{\rho},t)}{\partial t} = \mathcal{A}(\hat{\rho})x(\hat{\rho},t) + \mathcal{B}(\hat{\rho})u(t) + \mathcal{K}(\hat{\rho},t)e(\hat{\rho},t), \tag{1}$$

where  $x(\hat{\rho}, t)$  represents a collection of physical variables such as the poloidal magnetic flux profile  $\psi(\hat{\rho}, t)$  or the rotation profile  $V_{\phi}(\hat{\rho}, t)$ .

 $\mathcal{A}(\hat{\rho}), \mathcal{B}(\hat{\rho}), \text{ and } \mathcal{K}(\hat{\rho}) \text{ are infinite dimensional operators.}$ 

- To perform model Identification data is collected during high confinement (H-mode).
- The reference plasma state: Plasma current *I<sub>p</sub>* = 0.9 MA, 65% boot strap current, (H-mode): 3.5 < β<sub>N</sub> < 3.9 (β<sub>N</sub>: measure of pressure).
- Actuators modulated in open loop according to predefined waveforms around the values for the reference discharge.
- During each discharge one actuator is modulated while the other actuators are held constant and equal to the values for the reference discharge.

#### Actuators modulated to quantify plasma response

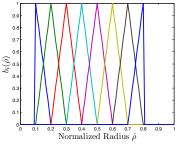


#### Discretization by Galerkin projection

- The infinite dimensional system (1) can be discretized by projecting the distributed variable x(ρ̂, t) onto a basis function space.
- The Galerkin projection reads

$$x(\hat{
ho},t) \approx \sum_{i=1}^{N} G_i(t) b_i(\hat{
ho}),$$
 (2)

• where  $b_i(\hat{\rho})$  are the basis functions. Piece-wise linear functions with i = 1, 2, ... N.



- The expansion coefficients,  $G_i(t)$ , called the Galerkin coefficients represent the model points.
- To determine the Galerkin coefficients, we multiply both sides of the expansion equation (2) with any basis function  $b_j(\hat{\rho}), j = 1, 2, ...N$  and integrate over the spacial coordinate to obtain,

$$\int_0^1 x(\hat{\rho}, t) b_j(\hat{\rho}) d\hat{\rho} = \int_0^1 \Big[ \sum_{i=1}^N G_i(t) b_i(\hat{\rho}) \Big] b_j(\hat{\rho}) d\hat{\rho}, \tag{3}$$

If the basis functions are orthonormal, i.e. ∫<sub>0</sub><sup>1</sup> b<sub>i</sub>(ρ̂)b<sub>j</sub>(ρ̂)dρ̂ = δ<sub>ij</sub>, then the coefficients G<sub>i</sub> can be computed explicitly. Otherwise the coefficients are obtained by solving a matrix equation.

After discretization we have a lumped parameter model, which reads:

$$\frac{dX(t)}{dt} = AX(t) + Bu(t) + Ke(t), \qquad (4)$$

where X(t) is the vector of Galerkin coefficients.

**Control Actuators** 

- 1. Co-current NBI
- 2. Counter-current NBI
- 3. Balanced NBI
- 4. Total ECRH and ECCD power from all the gyrotrons
- 5. Loop voltage

Then the model is fit to experimental data according to a least squares fit.

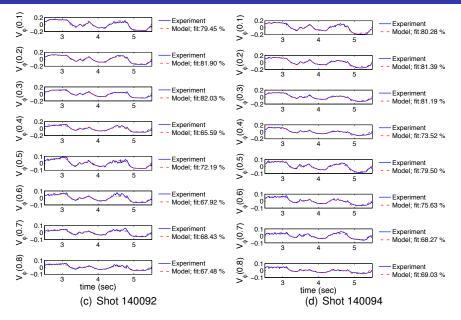
### Model identification

- System identification of the toroidal rotation profile carried out for 8 Galerkin coefficients computed at  $\hat{\rho} = 0.1, 0.2, ...0.8$ .
- Identification shots were organized into various groups; 1 group for little modulation and 1 group for each set of shots with a particular actuator modulated.
- The identification was then carried out in a step-wise procedure.
  - 1. Initial estimation of A was obtained using the group with little modulation
  - 2. Holding the slowest eigenmodes constant, the columns of the *B* matrix were estimated in subsequent steps, one column at at time.
  - 3. Each column estimated with the actuator corresponding to that column.
- The estimation process is carried out by fitting A and B to the data according to a least squares fit by minimizing the norm h(Q) = <sup>1</sup>/<sub>2</sub>tr(Q)

$$Q(\theta) = \frac{1}{N} \sum_{k=1}^{N} \epsilon(k, \theta) \epsilon^{T}(k, \theta)$$
(5)

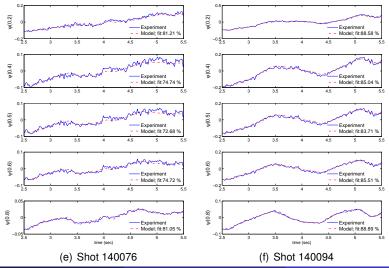
where  $\epsilon$  is the prediction error  $(X(k)|_{\text{measured}} - X(k)|_{\text{model}})$ , k is the sample, and  $\theta$  are the parameters to be determined.

### Fitted model describes rotation response accurately



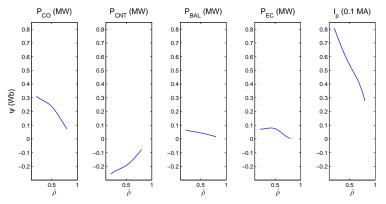
#### Fitted model describes $\psi$ response accurately

 The same identification process is carried out for the ψ profile (the poloidal flux relative to the boundary value)



## Steady state gains: magnetic profile $\psi$

• The estimated steady state gain matrix  $K_{sg} = -A^{-1}B$ 



- Steady state response of flux profile to unit change of the various inputs.
- The plasma current and co-current NBI are the most capable in adjusting the profile.
- The opposing affects of co-current and counter-current NBI are expected.

#### Two time scale coupled model

 The linearized coupled model between the magnetic profile ψ and a kinetic profile V<sub>φ</sub> can be written as

$$\frac{\partial \psi}{\partial t} = A_{11}\psi(t) + A_{12}V_{\phi}(t) + B_1u(t)$$
(6)

$$\epsilon \frac{\partial V_{\phi}}{\partial t} = A_{21}\psi(t) + A_{22}V_{\phi}(t) + B_2u(t)$$
(7)

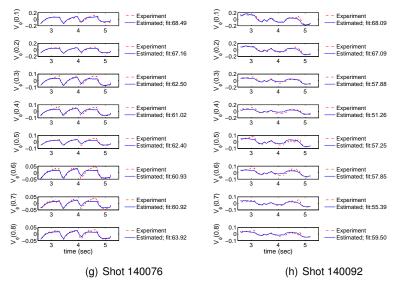
- $\epsilon$  is the ratio between energy confinement time and the characteristic resistive diffusion time ( $\epsilon \ll 1$ )
- In the limit  $\epsilon \rightarrow 0$  the model can be decomposed into slow and fast models of the form

$$\frac{\partial \psi}{\partial t} = A_{\text{slow}} \psi + B_{\text{slow}} u_{\text{slow}} \text{ and } V_{\phi}|_{\text{slow}} = C_{\text{slow}} V_{\phi}|_{\text{slow}} + D_{\text{slow}} u_{\text{slow}} (8)$$

$$\frac{\partial V_{\phi}|_{\text{fast}}}{\partial t} = A_{\text{fast}} V_{\phi}|_{\text{fast}} + B_{\text{fast}} u_{\text{fast}} (9)$$

## Two time scale model: $V_{\phi}|_{slow} + V_{\phi}|_{fast}$ , fitted model

• Using the previous model determined for  $\psi$  as  $A_{slow}$ ,  $B_{slow}$ 

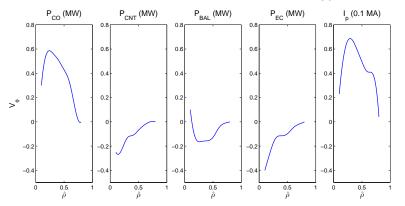


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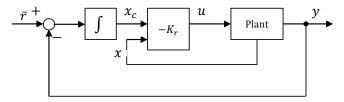
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### Static gains: rotation profile ( $V_{\phi}$ )

• The estimated steady state gain matrix  $K_{sg} = -C_{slow}A_{slow}^{-1}B_{slow} + D_{slow}$ 



## Optimal state feedback controller with integral action (Proportional + Integral control)



 To design the controller above, the plant is augmented with the integrator states x<sub>c</sub> = ∫ r̄ − y:

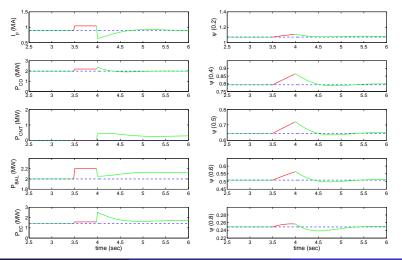
$$\dot{\bar{x}} = \begin{bmatrix} -C & 0\\ A & 0 \end{bmatrix} \bar{x} + \begin{bmatrix} -D\\ B \end{bmatrix} u + \begin{bmatrix} 1\\ 0 \end{bmatrix} \bar{r}$$
(10)

 Then using the augmented plant a simple state feedback control law of the form u(t) = -K<sub>r</sub>x̄(t) is determined to minimize the cost functional

$$J = E \left\{ \lim_{T \to \infty} \frac{1}{T} \int_0^T \left[ \bar{x}^T \bar{Q} \bar{x} + u^T \bar{R} u \right] dt \right\}$$
(11)

### Control simulation: successful disturbance rejection

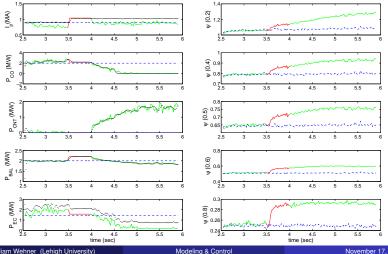
- Input disturbance at t = 3.5 s. The green period indicates the feedback is turned on and the red period indicates the feedback is turned off.
- The dash blue line is the target profile.



- Input disturbances are applied to all the actuators except the counter-current beam with a magnitude of about 10 – 15 % of their respective feedforward values.
- The feedback is turned off for 0.5 s to allow time for the disturbance to perturb the system.
- At *t* = 4.5 s the feedback is turned back on to regulate the states back to their reference trajectories.

### **Experimental result**

- Input disturbance at t = 3.5 s. The green period indicates the feedback is turned on and the red period indicates the feedback is turned off.
- (left) The black dots represent the delivered inputs, the green-red line ٩ represent the requested inputs. (Blue) target profile.



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- The experimental procedure is identical to the simulation.
- From 2.5 3.5 s the control performs well, holding the  $\psi$ -profile tight with the desired target.
- At 3.5 s an input disturbance is introduced and allowed to perturb the system for 0.5 s without feedback control (red portion).
- At 3.5 s the feedback is turned back on.
- Unfortunately, a mistake with the DIII-D settings disallowed the plasma current from going down.
- The neutral beams and gyrotrons are adjusted in the correct directions as predicted by the model, but the failure in *I*<sub>p</sub> actuation results in poor control.
- A second experimental attempt is scheduled for December 2011.

- Incorporate coupled evolution of toroidal rotation profile with the poloidal magnetic flux profile and the temperature profile.
- Use identified models in conjunction with Magnetohydrodynamic (MHD) stability models in development by Yongkyoon In (Fartech) to further study the stabilizing affects of the profile optimization on MHD instabilities.