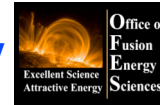


Supported by



Office of
Science



Supersonic gas jet fueling (XP 516)

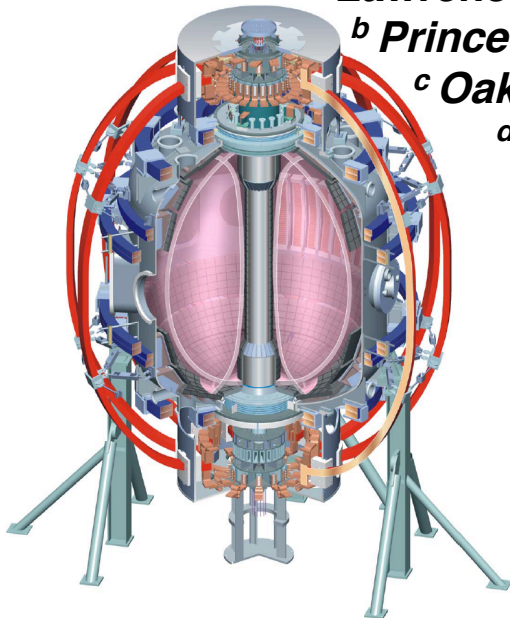
V.A. Soukhanovskii^a, H.W. Kugel^b, R. Kaita^b, A.L. Roquemore^b
with contributions from
M. Bell^b, W. Blanchard^b, C. Bush^c, R. Gernhardt^b,
G. Gettelfinger^b, T. Gray^b, R. Majeski^b, J. Menard^b,
T. Provost^b, R. Raman^d, P. Sichta^b

^a Lawrence Livermore National Laboratory, Livermore, CA, USA

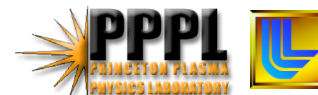
^b Princeton Plasma Physics Laboratory, Princeton, NJ, USA

^c Oak Ridge National Laboratory, Oak Ridge, TN, USA

^d University of Washington, Seattle, WA, USA



NSTX FY05 Results Review
12 December 2005
Princeton, NJ

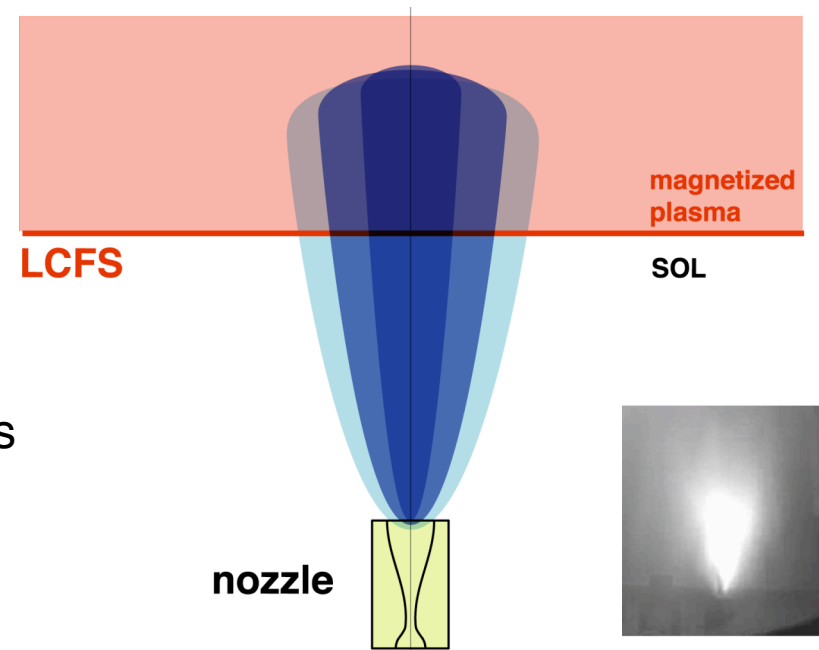


Supersonic gas jet is a unique fueling technique studied in NSTX

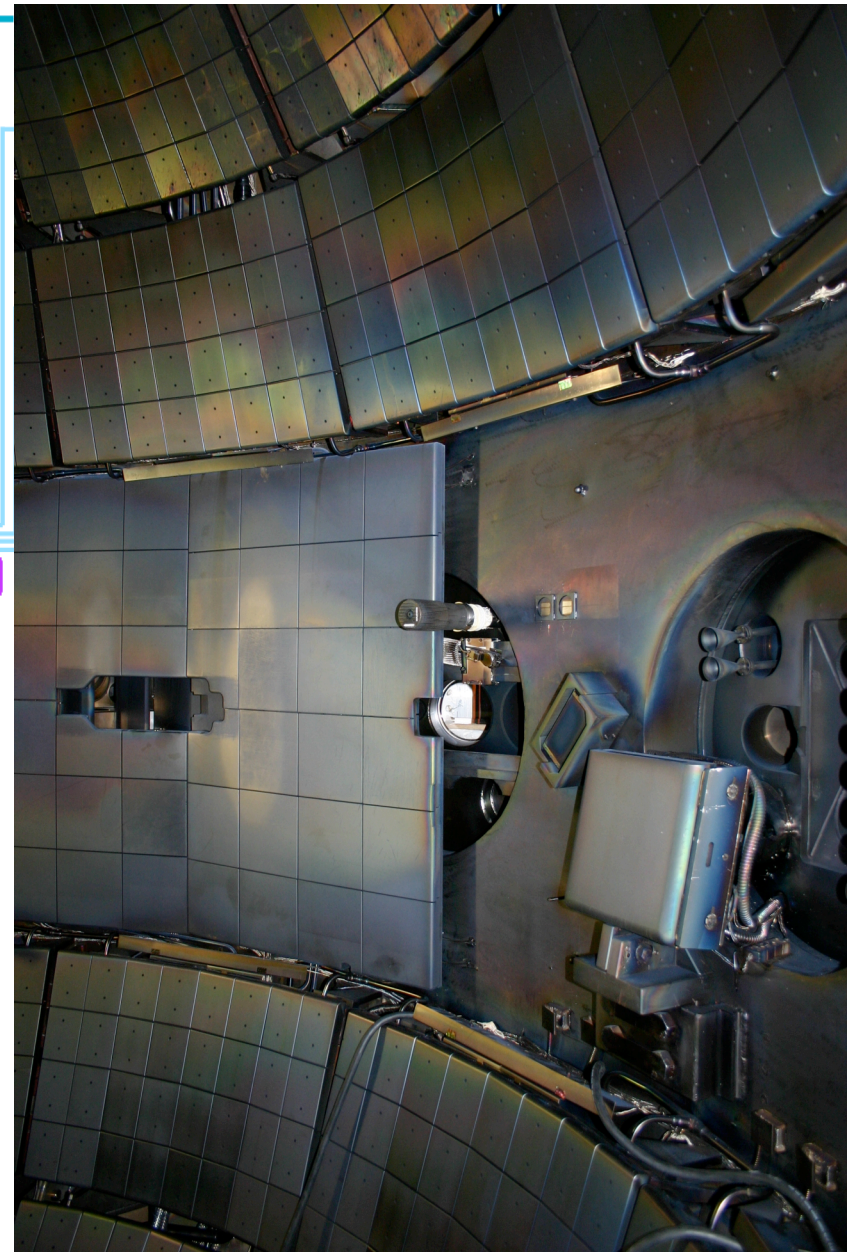
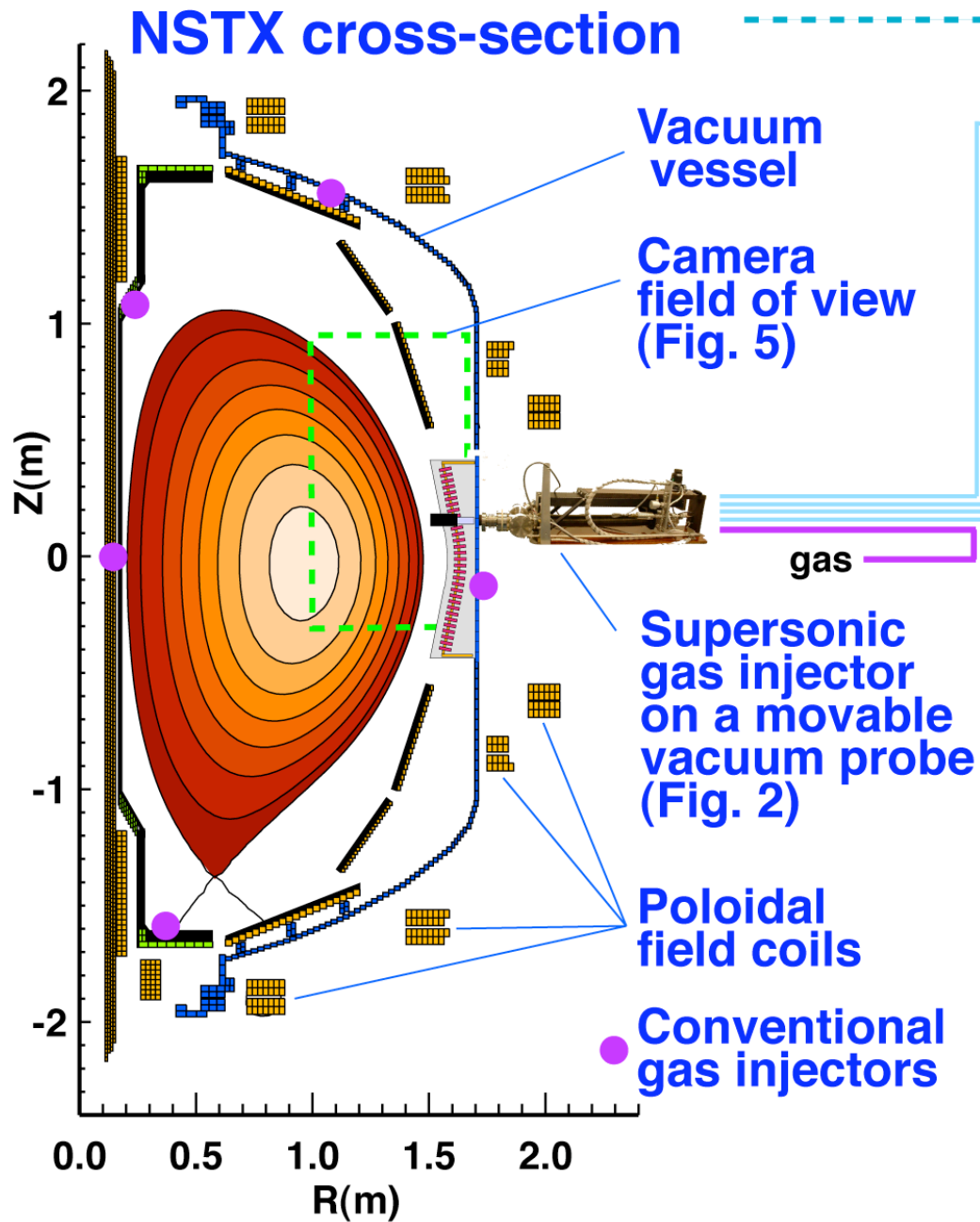
- **Improve / optimize gas fueling.** ITER will rely on central fueling (pellet, compact toroid), but plasma start-up and edge fueling will use gas puffing. **SJI is viewed as supplemental approach for ITER. Unique contribution of NSTX to novel fueling techniques.**
- Supersonic gas jet fueling results to date
 - limiter tokamaks (HL-1M, Tore Supra): injects 0.2-0.9 of total plasma inventory in several ms, perturbative, fueling efficiency 0.3-0.6
 - divertor tokamak (AUG), divertor stellarator (W7-AS): similar gas jet parameters, but FE \sim 0.1-0.3
- Implemented in **NSTX** in FY04, started XP 516 in FY05

Supersonic gas jet penetration mechanism

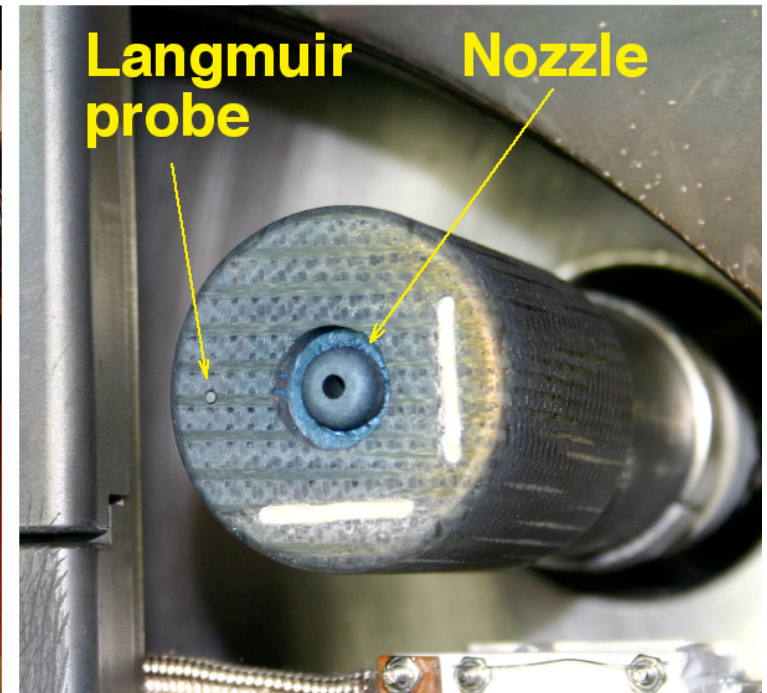
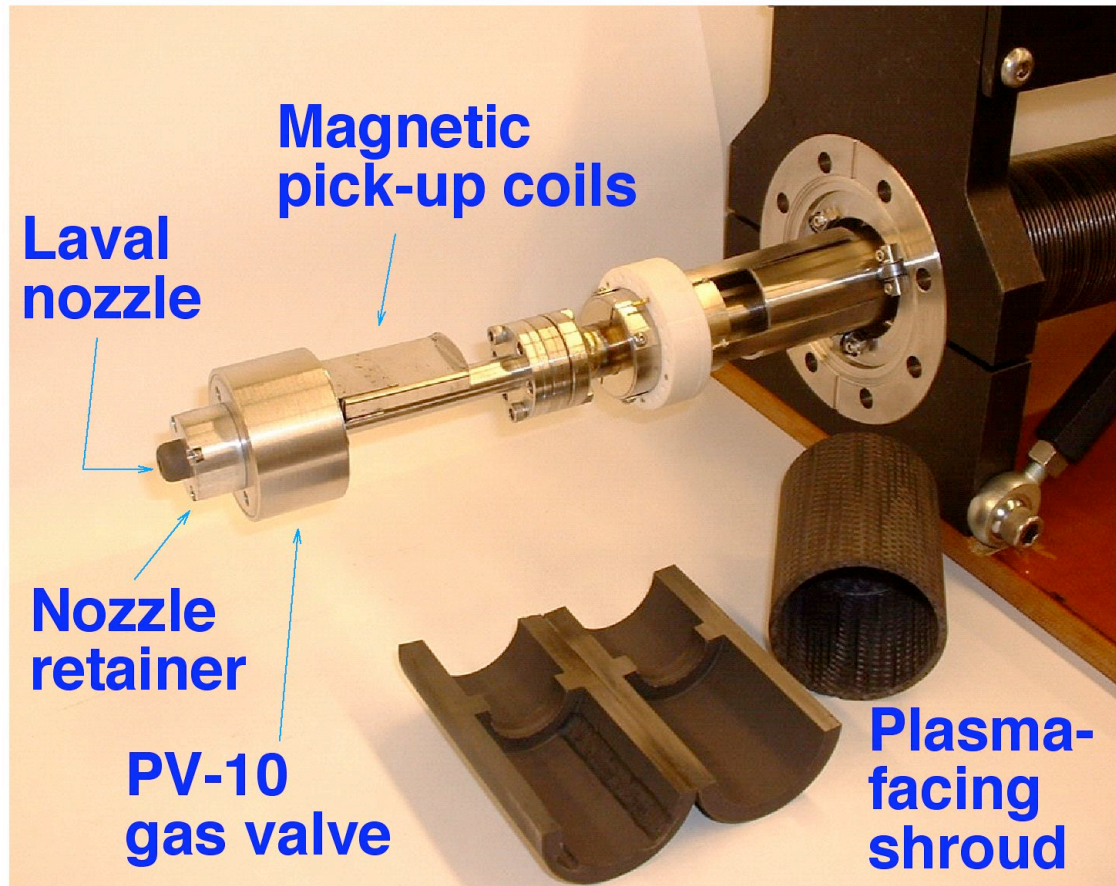
- Gas jet retains low-divergence shape (compressible fluid gasdynamics)
- Gas jet eventually ionizes and creates a plasmoid
- Gas jet retains cluster-molecular-atomic-ion structure
- SOL/edge electrons with low T_e do not fully penetrate gas jet
- Plasmoid can not penetrate deep into the magnetized plasma due to insufficient velocity and high plasma kinetic and magnetic pressure
- Single particle model is inapplicable
- Modeling must include continuity, momentum, energy balance (Braginskii) equations with detailed reaction rates and neutral transport (such as UEDGE+DEGAS 2)
- Velocity distribution function is drifting narrowed Maxwellian with $u_{\text{drift}} = u_{\text{flow}}$
- $u_{\text{flow}} = M c = M \sqrt{\gamma kT/m} > v_{\text{therm}}$



SGI on NSTX: placement and control elements

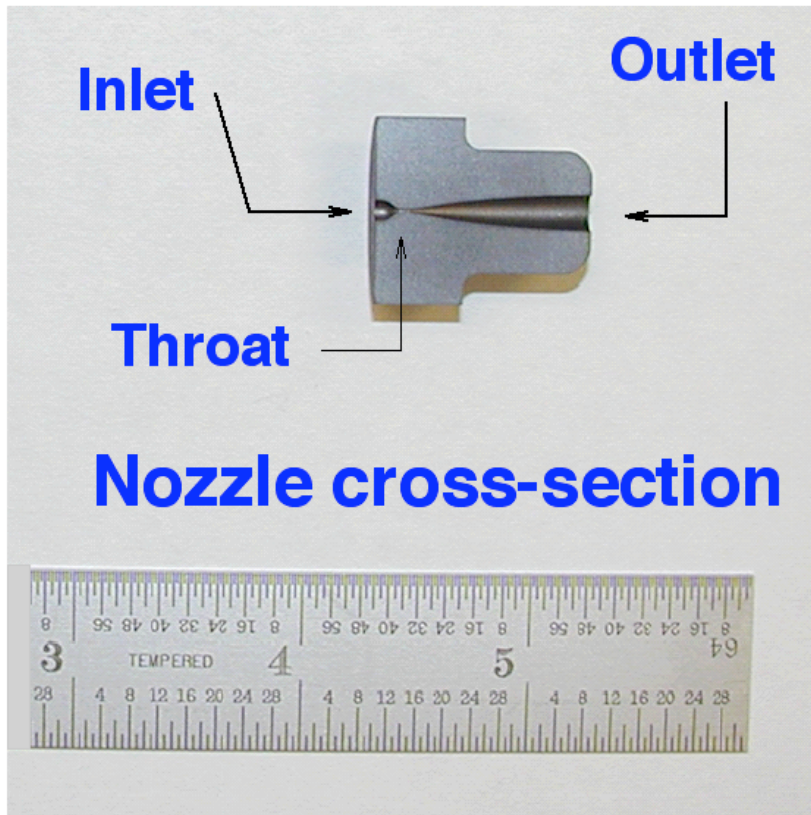


SGI head is a densely packed apparatus

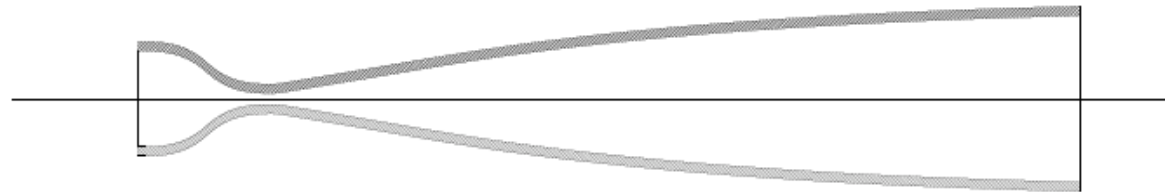


- Veeco PV-10 gas valve: $d_{throat}=0.02''$, typical opening time 1-2 ms, driving voltage 150 V
- Thermocouples in shroud and in gas valve
- Two magnetic pick-up coils on shroud front surface for B_z , B_t measurements
- Three magnetic pick-up coils in shielded box inside shroud for B_z , B_r and magnetic fluctuations measurement
- Langmuir probe: flush-mounted design, $d_{tip}=1.75$ mm, $I-V$ recorded at 5 kHz, $-50 < V < 50$
- Shroud: CFC and ATJ graphite

Laval contoured nozzle is used in NSTX SGI

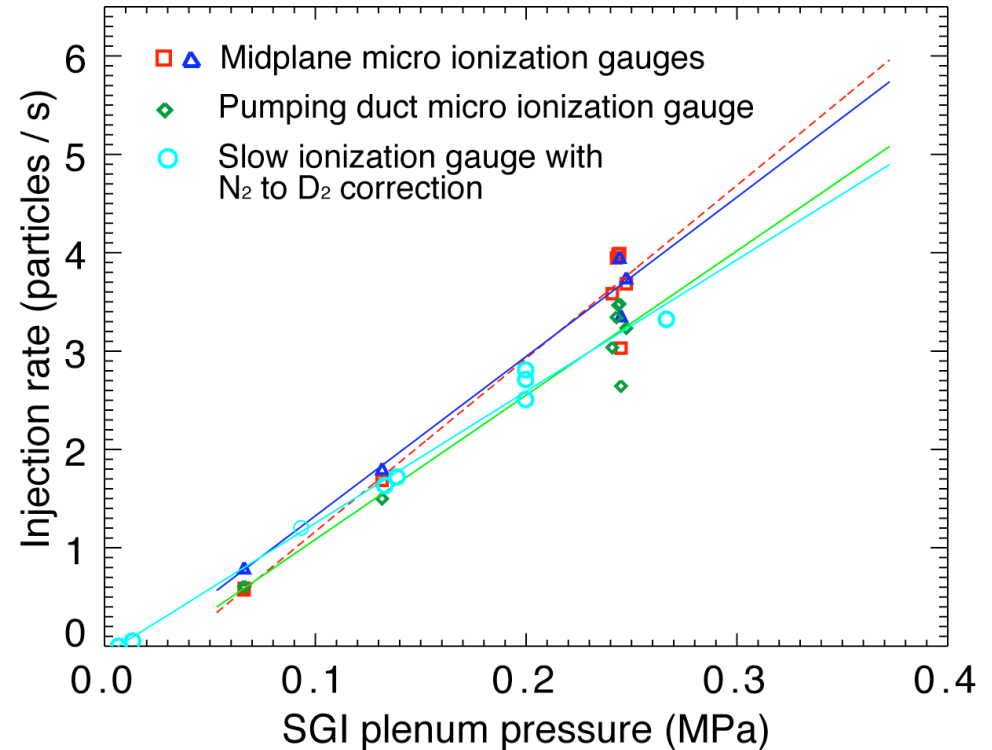
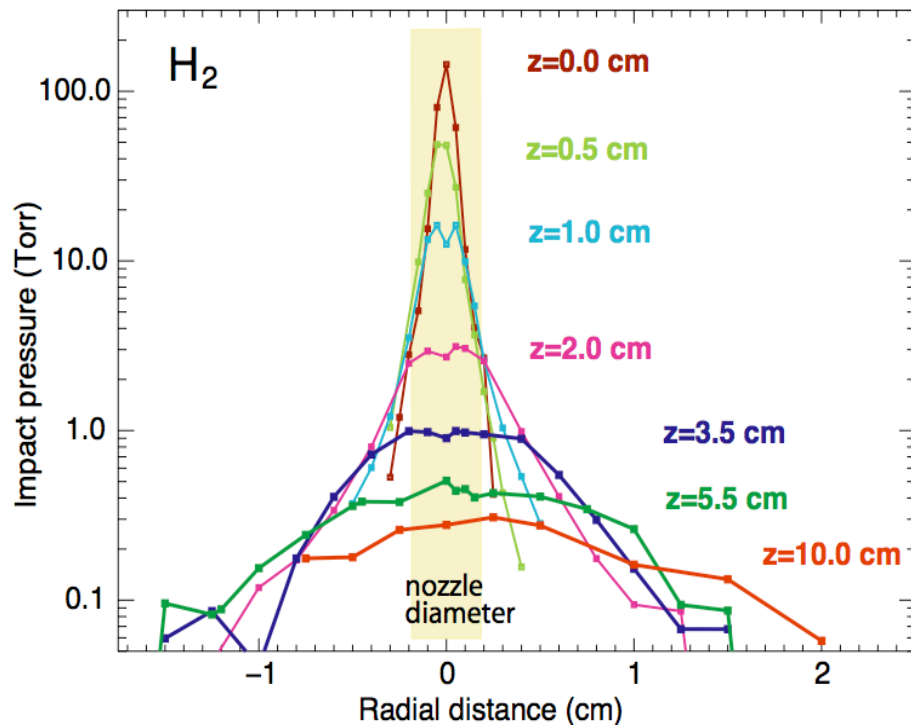


- Graphite nozzle $L = 23.4$ mm
- True Laval geometry calculated for air at $P=1$ atm, designed for $M = 8$, linearly scaled down to obtain $d_{throat} = 0.01''$ (throughput requirement)
- Compressible fluid theory: isentropic core and boundary layer scale differently!
- Nozzle is made by mechanical machining using special tool with tolerance $\pm 0.0025''$
- Nozzle attached to valve with a retainer using Viton O-ring



Nozzle design courtesy of Drs A. J. Smits, S. Zaidi (Princeton Univ.)

SGI parameters characterized off-line and *in situ*



- NSTX SGI is operated at 45-60 Torr l /s ($\sim (3.2 - 5) \times 10^{21}$ mol/s)
- NSTX gas injector rates: HFS: 10 - 50 Torr l /s, LFS: 20 - 120 Torr l /s
- Jet divergence half-angle: $6^\circ - 25^\circ$
- Hydrogen / Deuterium: $M = 4$, $T \sim 60 - 160$ K, $\rho \sim 5 \times 10^{17}$ cm^{-3} ,
 $Re = 6000$, $v_{therm} \sim 1100$ m/s, $v_{flow} = 2400$ m/s

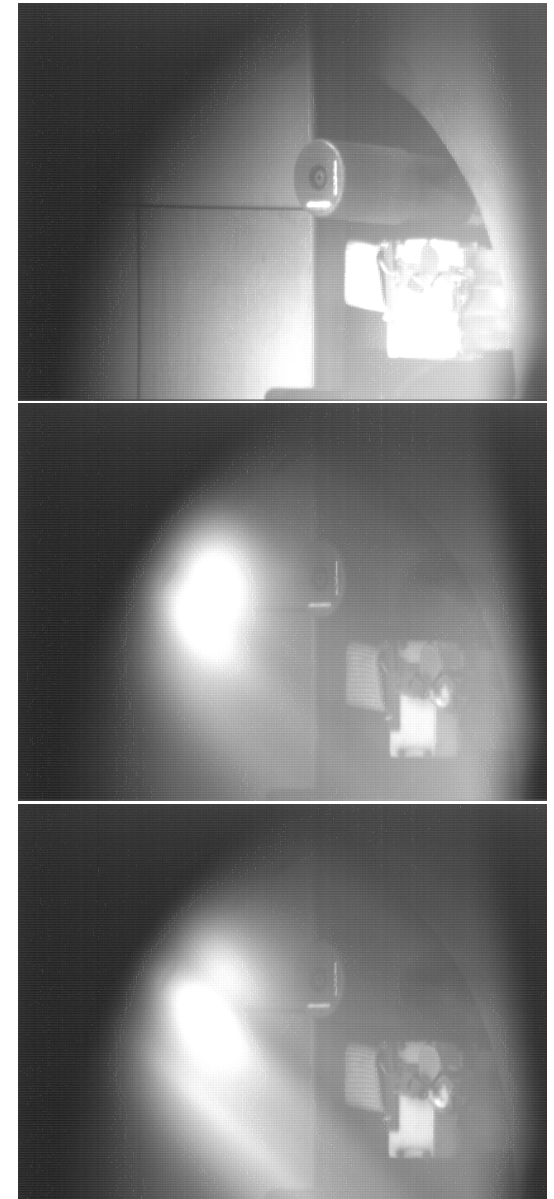
XP-516: Supersonic gas jet fueling

- Goals of XP:
 - Determine fueling efficiency (FE), penetration characteristics
 - Determine feasibility of fueling of H-mode plasmas
 - Study effects on edge plasmas
 - Develop model for high-pressure gas jet interaction with magnetized plasma

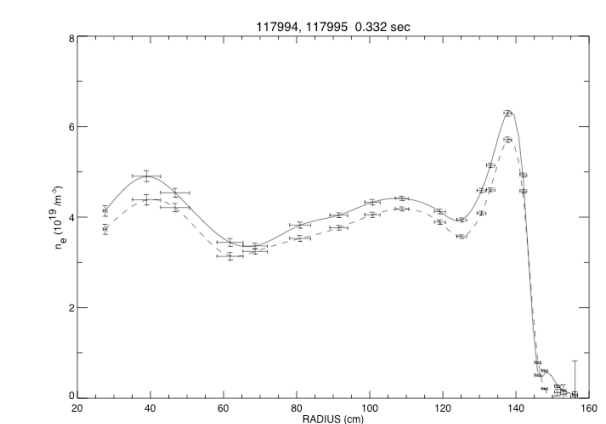
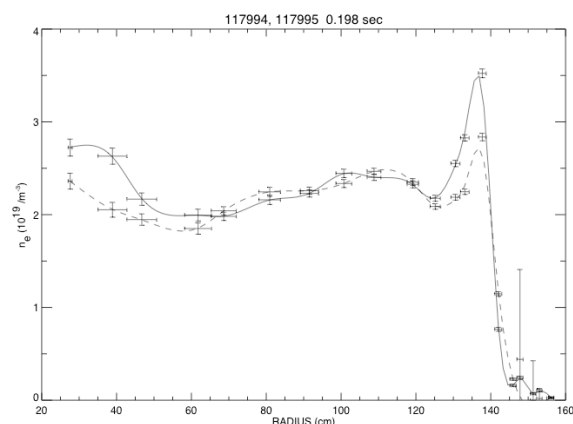
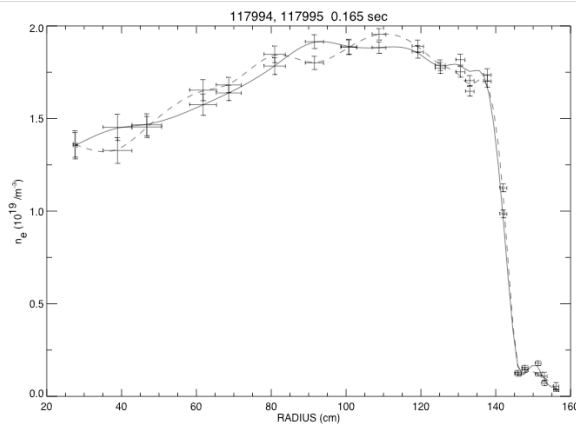
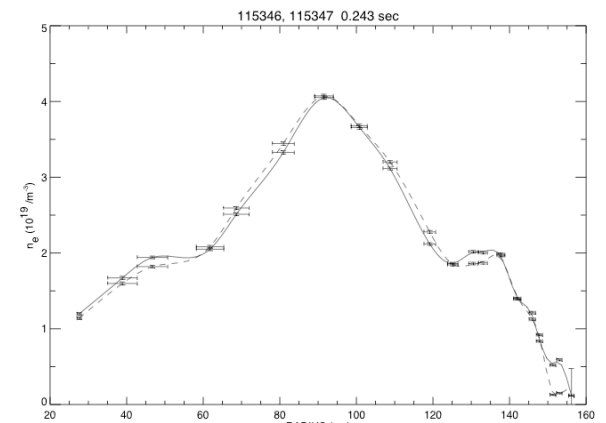
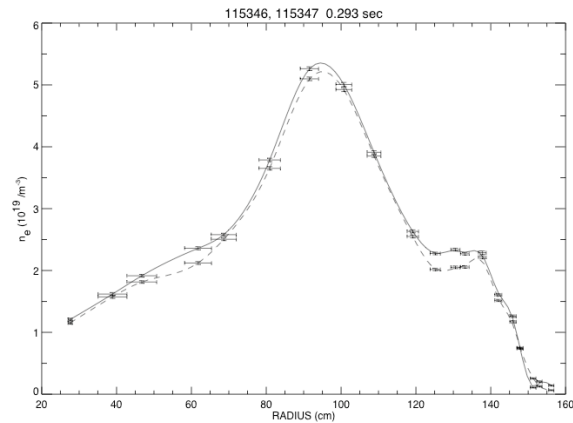
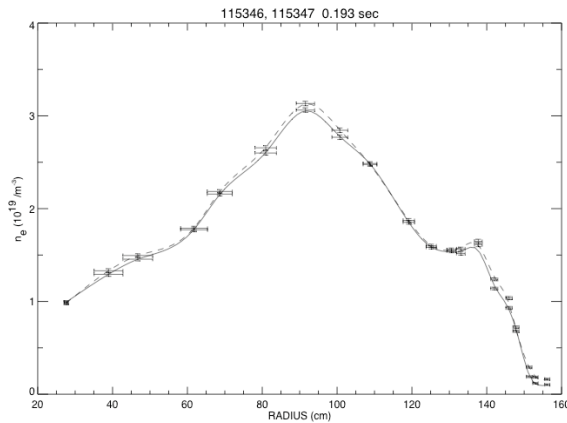
- SGI and “payload” diagnostics worked well in FY05:
 - Motorized SGI probe and PLC worked well
 - Diagnostic package (“payload”) commissioned and operated
 - Thermocouples measure room temperatures
 - Langmuir probe I_{sat} routinely obtained, T_e analysis in progress
 - Edge Magnetic Sensor: B_z , B_r , B_t coils and Mirnov coil signals routinely obtained, initial calibration completed, signals used in Poloidal Field only Plasma Start-up Experiment

XP 516 results

- LCFS - SGI distance scan in ohmic plasmas demonstrated that “**closer is better**”
- Cannot go closer than RF limiter in NBI-heated plasmas due to interaction with energetic particles
- **Compatible with H-mode** up to Inj. rate of 60 Torr l /s (conventional gas injection is not)
- **FE is between 0.1 and 0.3**
- SGI used gas pulses 70-300 ms
- Injected 0.1 - 1.0 of total plasma inventory
- Due to localization SGI gas pulse is not seen as pulse on MPTS or interferometry n_e traces
- Need to improve FE analysis: Present analysis for N_e only, need to exclude carbon contribution to particle inventory (using N_i from CHERS)



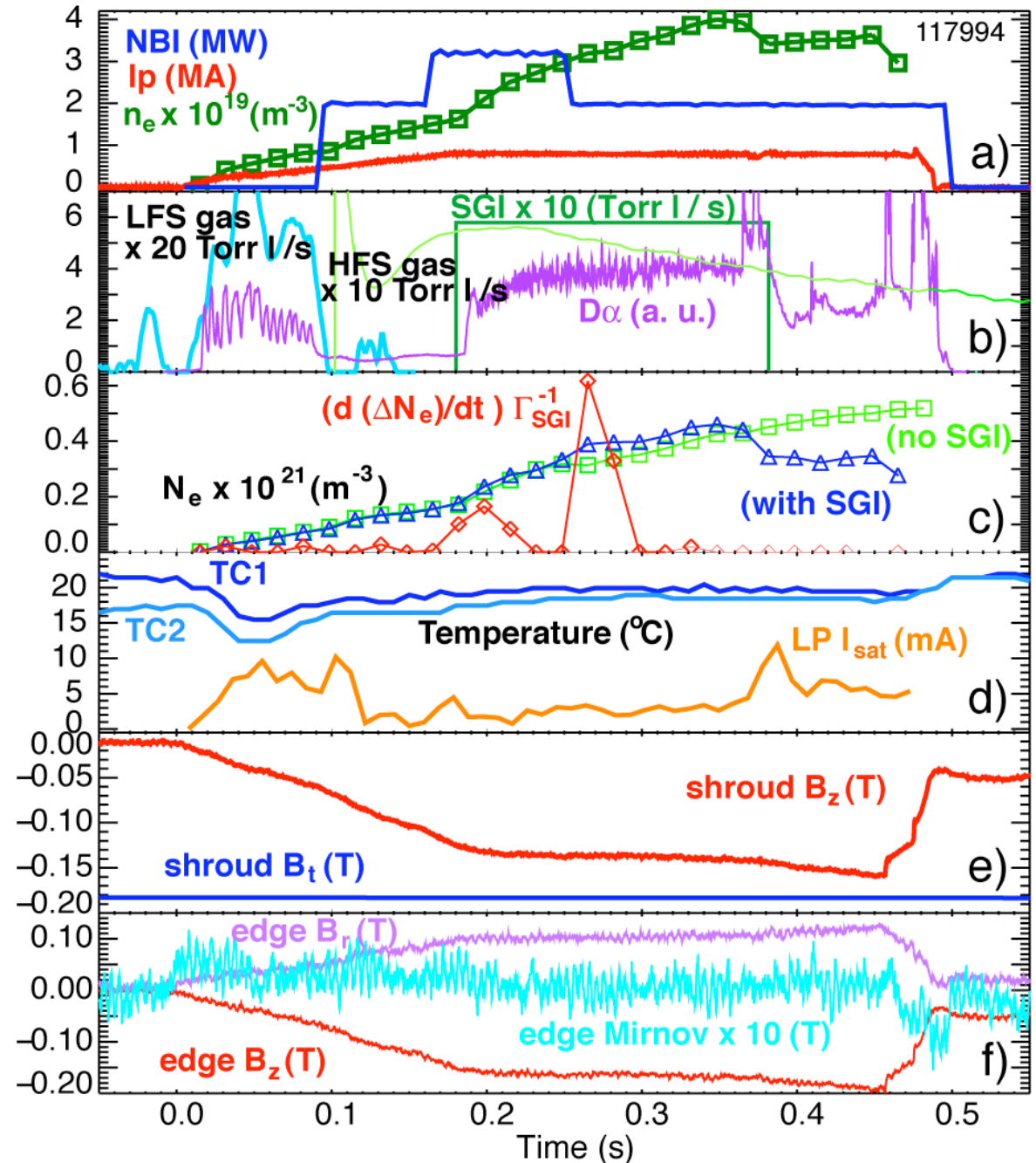
Particles from SGI are deposited at the edge



- SGI increases edge density (in H-modes, “ears” density)
- Effect is not always large -> need to look at total particle inventory
- Edge MHD and ELMs affect fueling

SGI and diagnostics perform well in real plasmas

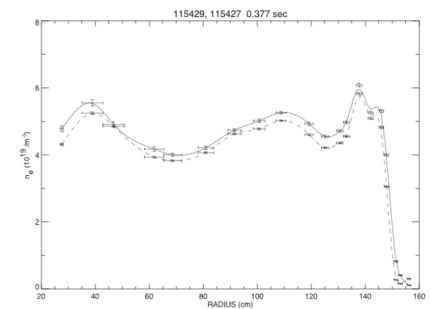
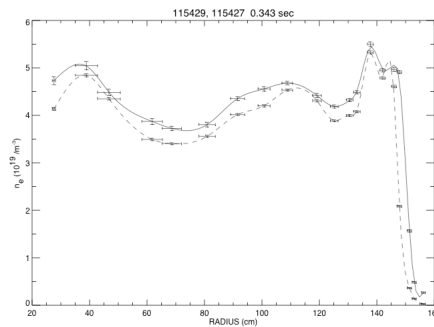
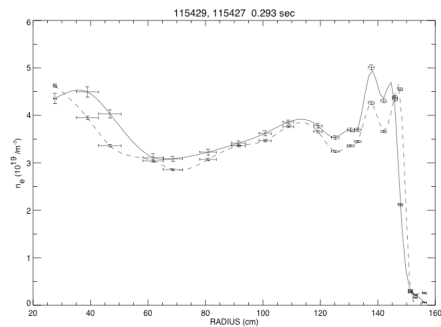
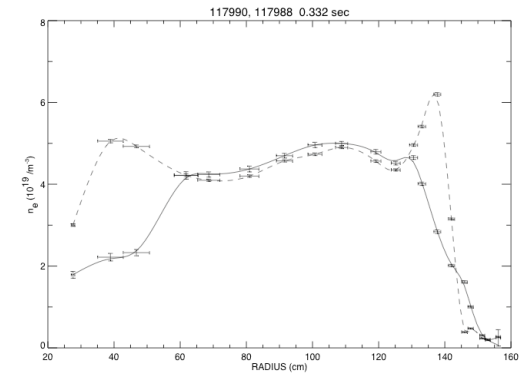
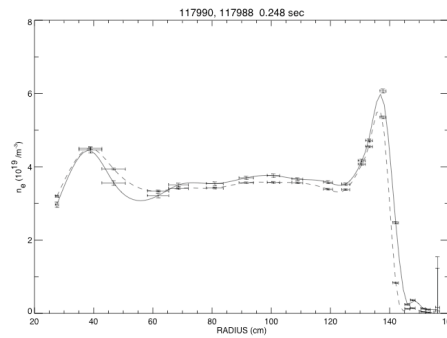
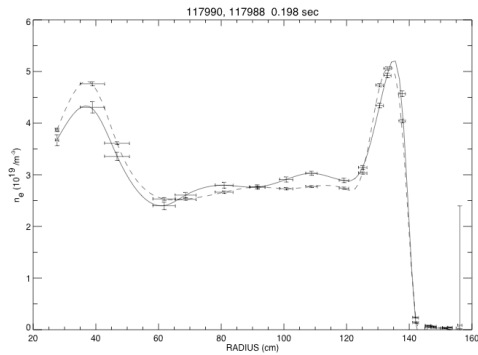
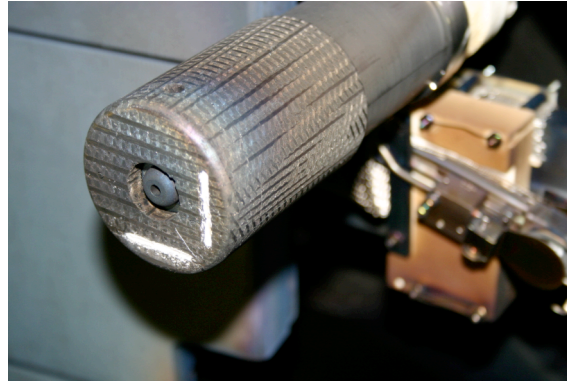
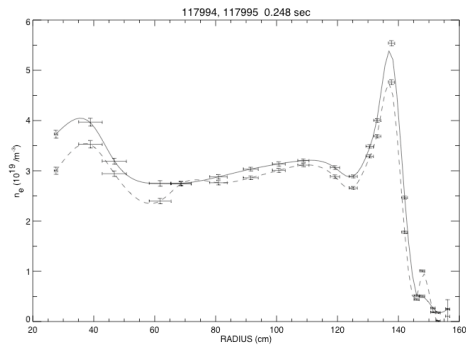
- Example: NSTX 2 MW NBI-heated H-mode plasma
- SGI starts at 0.180 s
- SGI rate ~ 55 Torr l / s
- FE $\sim 0.1 - 0.3$
- Good diagnostic signal SNR



Summary

- Supersonic gas injector and diagnostic package on a movable probe commissioned and operated on NSTX in FY04 - FY05
- Supersonic gas jet used to fuel ohmic and 2-6 MW NBI-heated L- and H-mode plasmas. Compatibility with H-mode pedestal has been demonstrated.
- Measured fueling efficiency 0.1 - 0.3
- Supersonic gas jet does not perturb plasma edge
- Need to finish XP 516 (fueling during plasma start-up, H-mode access with SGI, ...)
- Expected to be an important tool in lithium experiments

Backup slides



NSTX reference data

NSTX fueling

- Gas injection: low field side (LFS, top + side), high field side (HFS, midplane + shoulder), private flux region. D_2 , He, injected at $S = 20 - 120$ Torr l / s.
- Neutral beam injection system: three beams, 40 - 100 keV, 6 MW, fueling rate: $S < 6$ Torr l / s
- Supersonic gas injection: $S = 60$ Torr l / s

NSTX wall conditioning

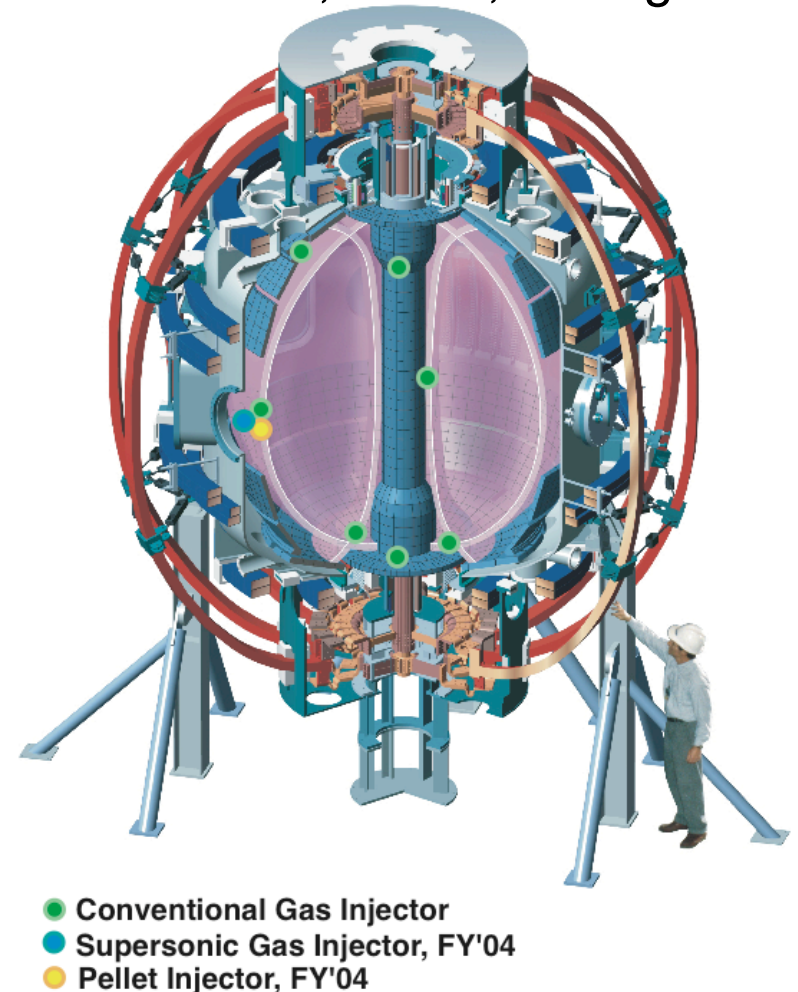
- Between shots He GDC
- He conditioning plasmas
- TMB and Plasma TMB

NSTX pumping

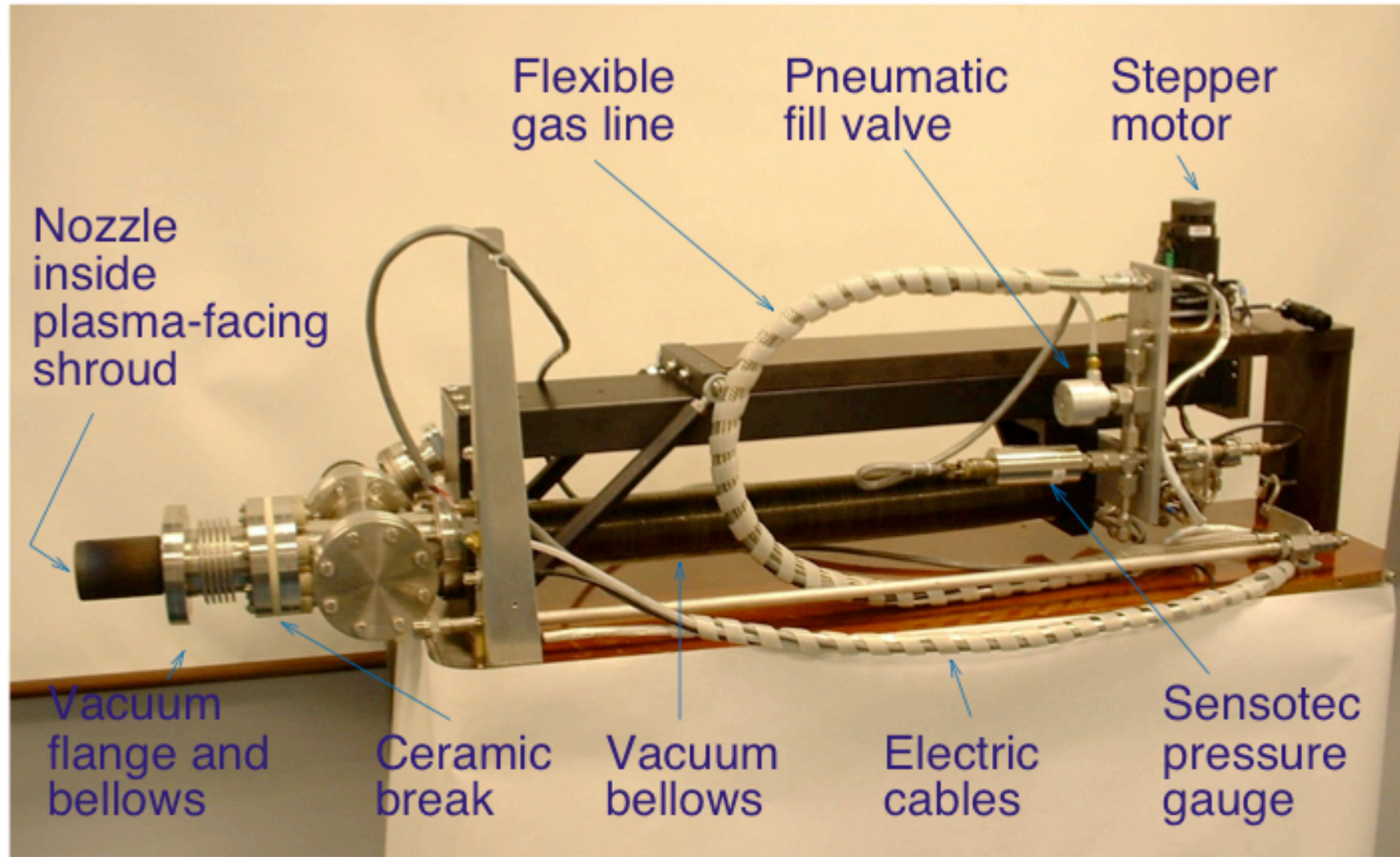
- Turbomolecular pump (3400 l / s)
- NBI cryopump (50000 l / s)
- Conditioned walls

PFC

- ATJ graphite tiles on divertor and passive plates
- ATJ and CFC tiles on center stack
- Tile thickness 1" and 2"

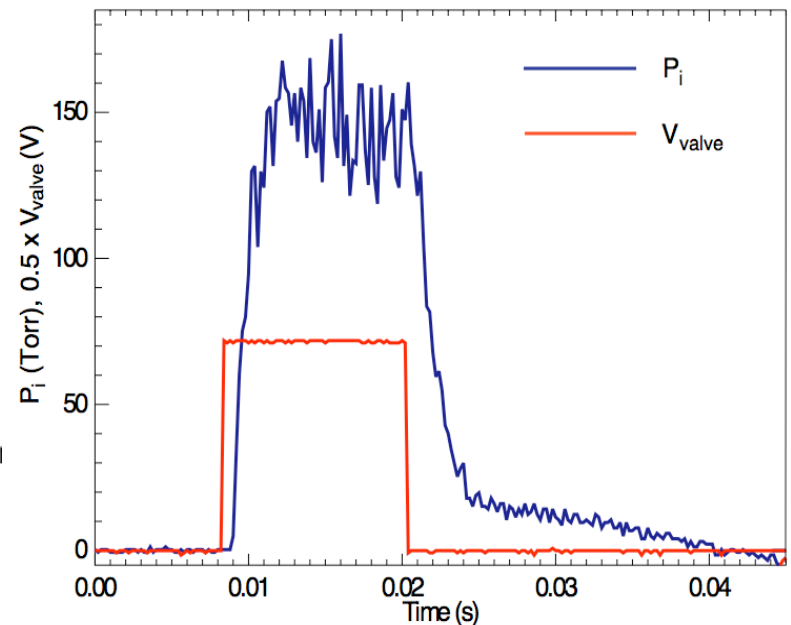
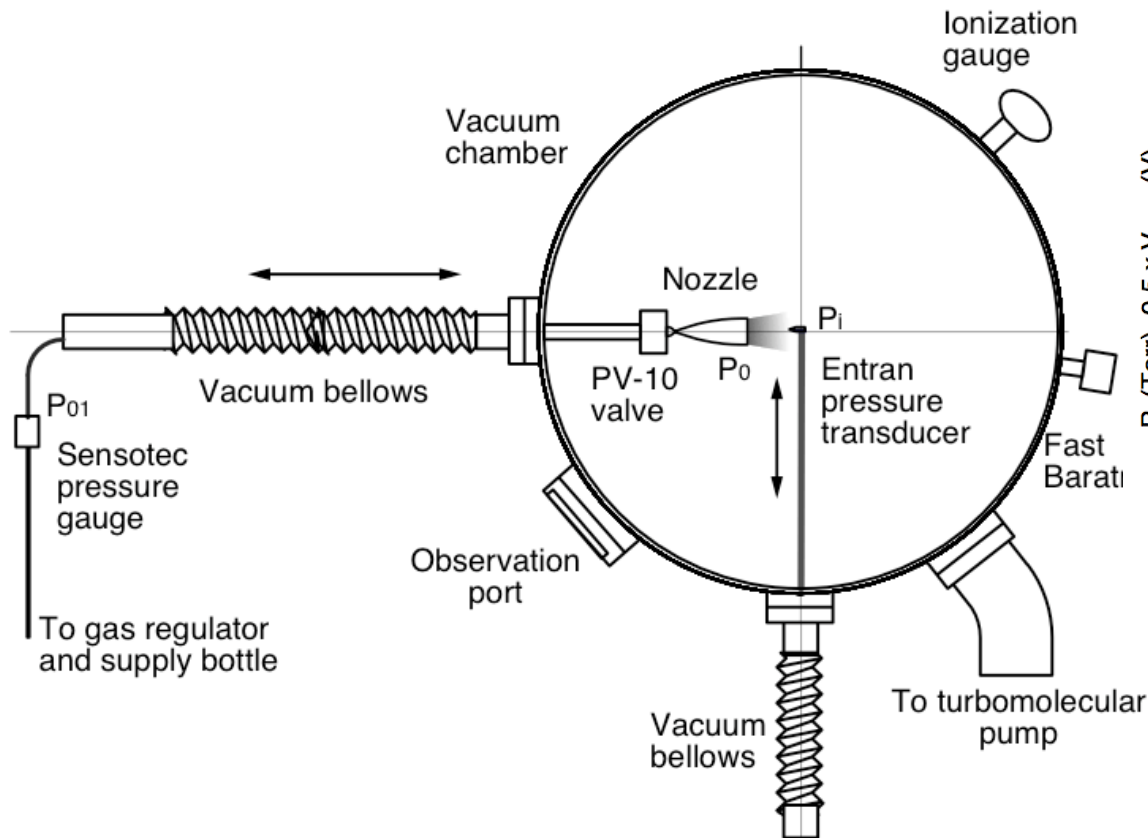


SGL: fueling and diagnostic packages on a probe



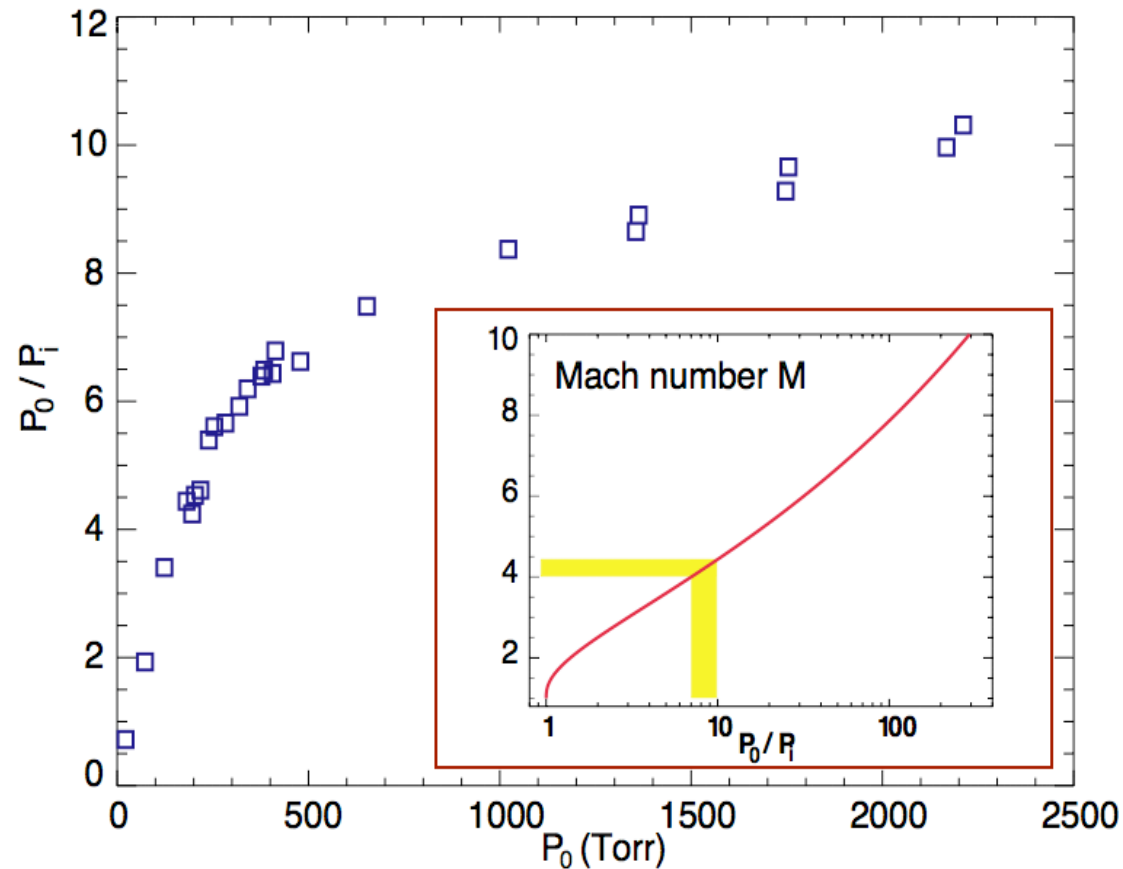
- Thermionics ZC-450 movable probe, stroke 24", travel rate < 15 in/s
- 6-axis Stepper Motor Controller, controlled using RS-232 port
- EPICS software used for SMC control and communication with Vacuum PLC

Laboratory tests designed to evaluate SGI



- Any nozzle must be tested to enable comparison with calculations - real nozzles often do not perform as expected
- Flow parameters diagnostic methods: Shadowgraphy, Schlieren photography, Laser induced fluorescence, Electron beam fluorescence, Laser scattering, Dust imaging, and others **are either too complicated or would not work in vacuum, in a pulsed regime**
⇒ Impact pressure measurement + supersonic Rayleigh-Pitot law for Mach number and jet pressure profile measurements at various distances z from the nozzle

Laboratory tests confirm high Mach number

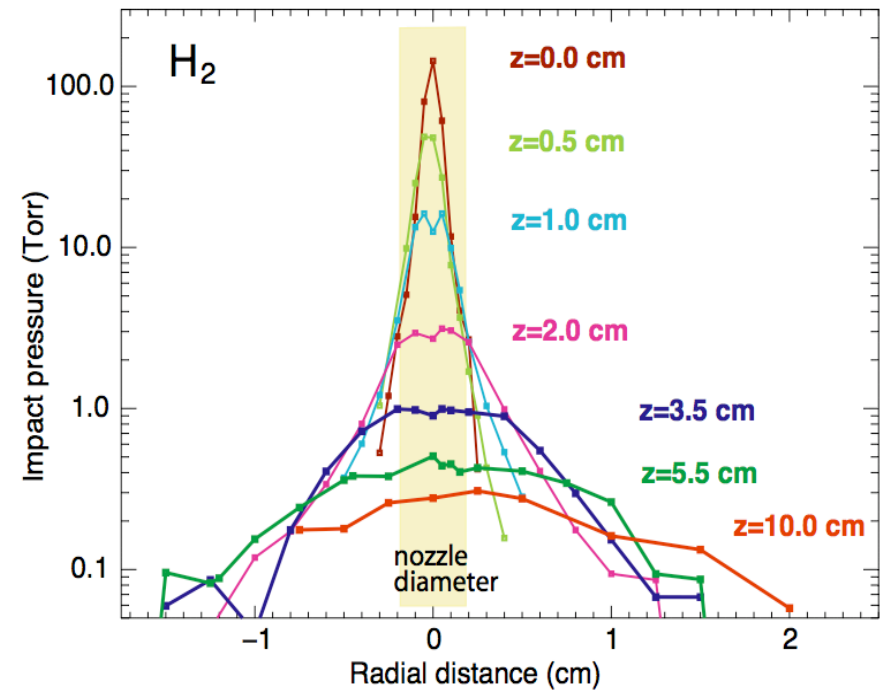
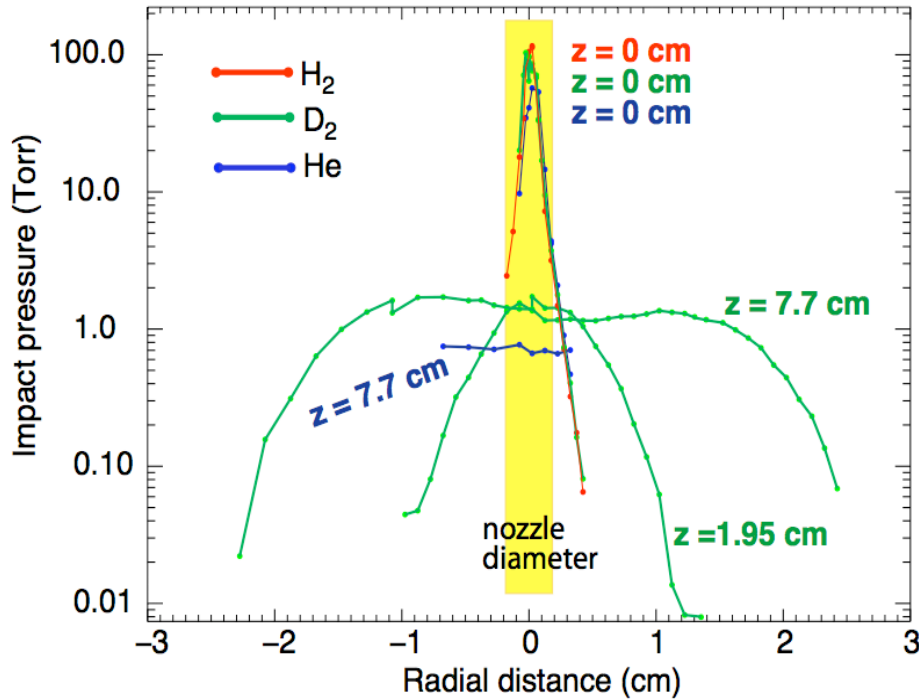


(gauge)

- Supersonic Rayleigh-Pitot law used to infer Mach number M from P_0/P_i measurements

$$\frac{P_i}{P_0} = \left(\frac{(\gamma + 1) M^2}{(\gamma - 1) M^2 + 2} \right)^{\gamma/(\gamma-1)} \left(\frac{\gamma + 1}{2\gamma M^2 - (\gamma - 1)} \right)^{1/(\gamma-1)}$$

Laboratory tests confirm low divergence of gas jet



Jet divergence half-angle: $6^\circ - 25^\circ$

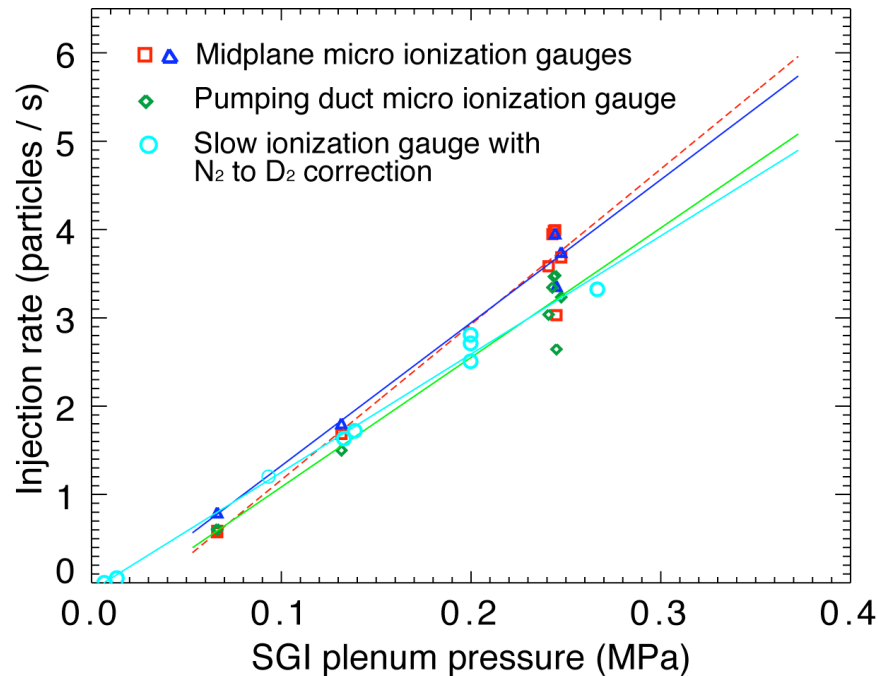
D_2 : $M = 4$, $T \sim 60 - 160 \text{ K}$, $\rho \sim 5 \times 10^{17} \text{ cm}^{-3}$, $Re = 6000$

D_2 : $v_{therm} \sim 1100 \text{ m/s}$, $v_{flow} = 2400 \text{ m/s}$

$$u_{max} = \sqrt{\frac{2\gamma}{\gamma-1} \frac{kT_0}{m}}$$

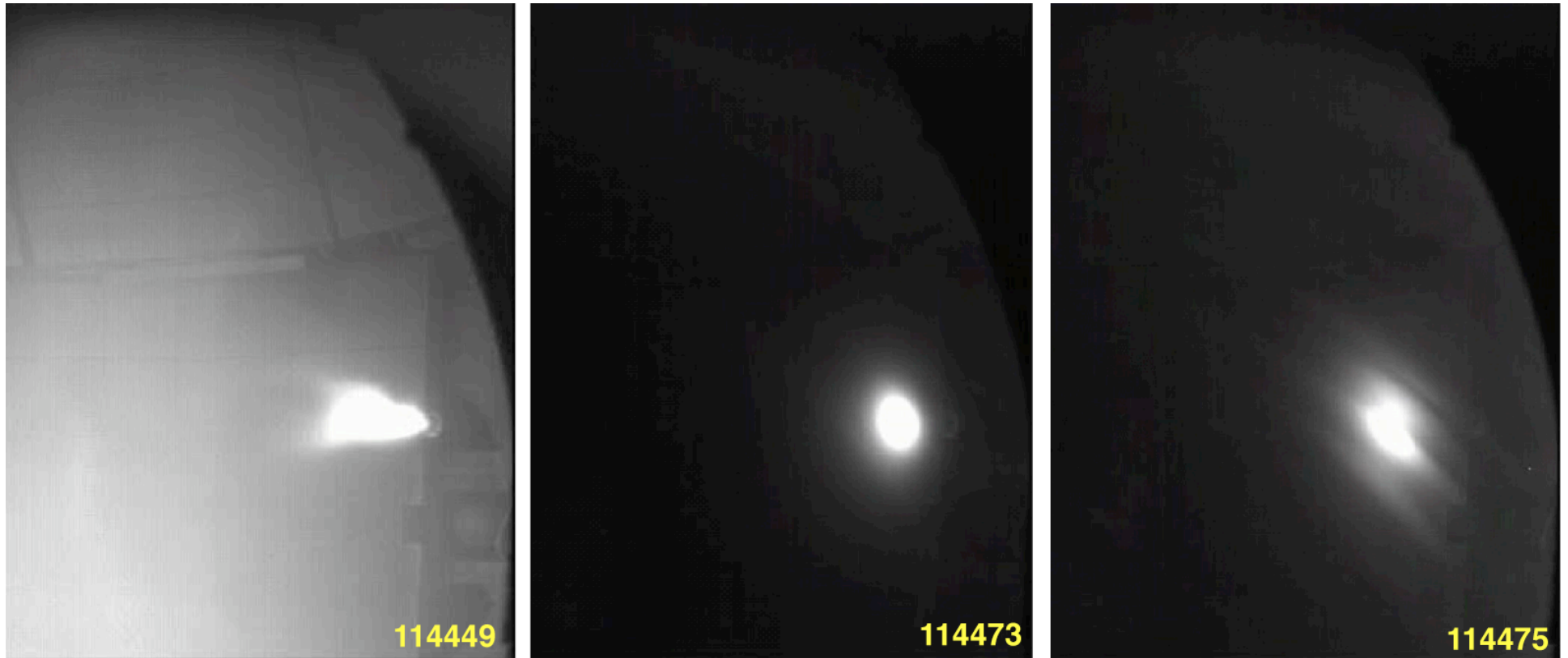
For comparison: divergence half-angle of viscous sonic flow - 30° ,
molecular effusion from orifice - 60°

Flow rate is measured *in situ* on NSTX



- Flow rate (Torr l / s): $\Gamma = V_{NSTX} dP / dt$
- NSTX SGI is operated at 45-60 Torr l / s ($\sim (3.2 - 5) \times 10^{21}$ mol/s)
- Future SGI may require $P_{plenum} > 2500$ Torr
- NSTX gas injector rates: HFS: 10 - 50 Torr l / s, LFS: 20 - 120 Torr l / s

Fast imaging of gas jet - plasma interaction



- Used Canadian Photonic camera with 0.5-2 ms framing rate
- Example frames above: (a) collapsing plasma with a wide $T_e = 3$ eV, $n_e = (2-2.5) \times 10^{18} \text{ m}^{-3}$ scrape-off layer, (b) 6 MW NBI-heated L-mode plasmas and (c) 4 MW NBI-heated H-mode plasmas