

# Performance Assessment of Model-Based Optimal Feedforward and Feedback Current Profile Control in NSTX-U using the TRANSP Code

Zeki Ilhan<sup>1</sup>, William P. Wehner<sup>1</sup>, Eugenio Schuster<sup>1</sup>,  
Mark D. Boyer<sup>2</sup>, David A. Gates<sup>2</sup>, Stefan P. Gerhardt<sup>2</sup>, Jonathan E. Menard<sup>2</sup>

<sup>1</sup>Department of Mechanical Engineering & Mechanics  
Lehigh University

<sup>2</sup>Princeton Plasma Physics Laboratory

*zeki\_ilhan@lehigh.edu*

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**ENERGY** | Office of  
Science



# Motivation for Current Density Profile Control in NSTX-U

- There is growing consensus that the path to an economical power-producing reactor is the “Advanced Tokamak” (AT) concept.
- AT operational goals for the NSTX-U include [1]:
  - Non-inductive sustainment of the high- $\beta$  spherical torus. (Fusion power scales as  $P_{fus} \approx \beta^2 B^4$ )
  - High performance equilibrium scenarios with neutral beam heating.
  - Longer pulse durations.
- Active, model-based, feedback control of the current density profile evolution can be useful to achieve these AT operational goals.
- Relation between  $\iota$ -profile and the toroidal current density ( $j_\phi$ ) profile [2]:

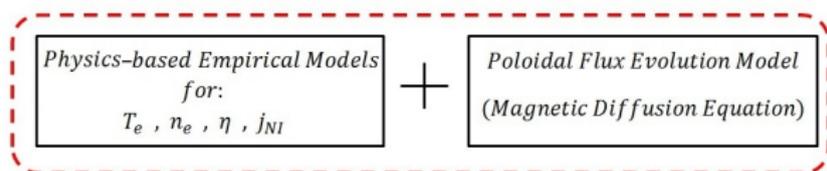
$$\iota(\hat{\rho}, t) = \frac{R_0 \mu_0}{\hat{\rho}^2 B_\phi} \int_0^{\hat{\rho}} j_\phi(\hat{\rho}', t) \hat{\rho}' d\hat{\rho}' = \underbrace{-\frac{d\Psi}{d\Phi}}_{-(\partial\psi/\partial\hat{\rho})/B_{\phi,0}\rho_b^2\hat{\rho}}$$

- Control of the  $\iota$ -profile is equivalent to control of the current density profile,  $j_\phi(\hat{\rho}, t)$ , and the control of the poloidal flux gradient profile,  $\partial\psi/\partial\hat{\rho}$ .

[1] GERHARDT, S. P., ANDRE, R., and MENARD, J. E., Nuclear Fusion **52** (2012).

[2] J. Wesson, *Tokamaks* (Oxford University Press, 3rd edition, 2004).

# First-Principles-Driven (FPD) Current Profile Modeling



## First – Principles – Driven (FPD) Current Profile Evolution Model

- The evolution of the **poloidal magnetic flux** is given by the **Magnetic Diffusion Equation [3]**

$$\frac{\partial \psi}{\partial t} = \frac{\eta(T_e)}{\mu_0 \rho_b^2 \hat{F}^2} \frac{1}{\hat{\rho}} \frac{\partial}{\partial \hat{\rho}} \left( \hat{\rho} D_\psi \frac{\partial \psi}{\partial \hat{\rho}} \right) + R_0 \hat{H} \eta(T_e) \frac{\langle \bar{j}_{NI} \cdot \bar{B} \rangle}{B_{\phi,0}}, \quad (1)$$

with boundary conditions

$$\left. \frac{\partial \psi}{\partial \hat{\rho}} \right|_{\hat{\rho}=0} = 0, \quad \left. \frac{\partial \psi}{\partial \hat{\rho}} \right|_{\hat{\rho}=1} = -\frac{\mu_0}{2\pi} \frac{R_0}{\hat{G} \Big|_{\hat{\rho}=1} \hat{H} \Big|_{\hat{\rho}=1}} I(t), \quad (2)$$

where  $D_\psi(\hat{\rho}) = \hat{F}(\hat{\rho}) \hat{G}(\hat{\rho}) \hat{H}(\hat{\rho})$ , and  $\hat{F}$ ,  $\hat{G}$ ,  $\hat{H}$  are geometric factors pertaining to the magnetic configuration of a particular equilibrium.

[3] OU, Y., LUCE, T. C., SCHUSTER E. et al., *Fusion Engineering and Design* (2007).

# First-Principles-Driven (FPD) Current Profile Modeling

- NSTX-U-tailored [4] empirical models [5] for the electron temperature, electron density, plasma resistivity, and noninductive current drives [6] take the form

$$n_e(\hat{\rho}, t) = n_e^{prof}(\hat{\rho})u_n(t) \quad (3)$$

$$T_e(\hat{\rho}, t) = k_{T_e}(\hat{\rho}, t_r) \frac{T_e^{prof}(\hat{\rho}, t_r)}{n_e(\hat{\rho}, t)} I(t) \sqrt{P_{tot}(t)} \quad (4)$$

$$\eta(T_e) = \frac{k_{sp}(\hat{\rho}, t_r) Z_{eff}}{T_e(\hat{\rho}, t)^{3/2}} \quad (5)$$

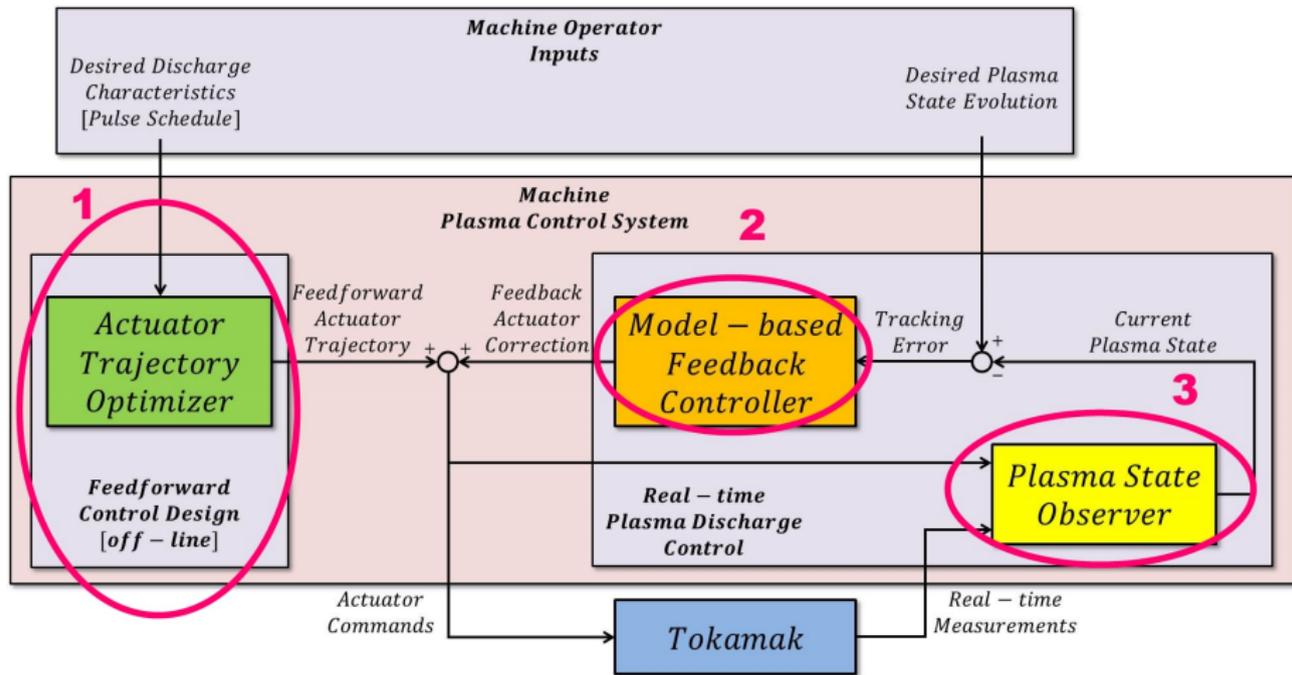
$$\begin{aligned} \frac{\langle \bar{\mathbf{j}}_{ni} \cdot \bar{\mathbf{B}} \rangle}{B_{\phi,0}} &= \sum_{i=1}^6 \frac{\langle \bar{\mathbf{j}}_{nbi_i} \cdot \bar{\mathbf{B}} \rangle}{B_{\phi,0}} + \frac{\langle \bar{\mathbf{j}}_{bs} \cdot \bar{\mathbf{B}} \rangle}{B_{\phi,0}} \\ &= \sum_{i=1}^6 k_i^{prof}(\hat{\rho}) J_i^{dep}(\hat{\rho}) \frac{\sqrt{T_e(\hat{\rho}, t)}}{n_e(\hat{\rho}, t)} P_i(t) \\ &\quad + \frac{k_{JeV} R_0}{\hat{F}(\hat{\rho})} \left( \frac{\partial \psi}{\partial \hat{\rho}} \right)^{-1} \left[ 2\mathcal{L}_{31} T_e \frac{\partial n_e}{\partial \hat{\rho}} + \{2\mathcal{L}_{31} + \mathcal{L}_{32} + \alpha \mathcal{L}_{34}\} n_e \frac{\partial T_e}{\partial \hat{\rho}} \right] \end{aligned} \quad (6)$$

[4] ILHAN, Z. O. et al., 55<sup>th</sup> Annual Meeting of the APS DPP (2013)

[5] BARTON, J. E. et al., 52<sup>nd</sup> IEEE CDC (2013)

[6] SAUTER, O. et al., Physics of Plasmas (1999), (2002)

# Possible Uses of the FPD Model



Schematics of the plasma profile control system

# Feedforward Actuator Trajectory Optimization

- **Objective:** Design the actuator trajectories that can steer the plasma to a target state characterized by the safety factor profile  $q^{tar}(\hat{\rho}, t_f)$  or rotational transform profile  $\iota^{tar}(\hat{\rho}, t_f)$  at a specified time  $t_f$  during the discharge such that the achieved plasma state is as stationary in time as possible.
- **Cost functional** defined as:

$$J(t_f) = k_q J_q(t_f) + k_{ss} J_{ss}(t_f)$$

where  $k_{ss}$  and  $k_q$  are the weight factors representing the relative importance of the plasma state characteristics and

$$J_q(t_f) = \int_0^1 W_q(\hat{\rho}) [q^{tar}(\hat{\rho}) - q(\hat{\rho}, t_f)]^2 d\hat{\rho} \quad (7)$$

$$J_{ss}(t_f) = \int_0^1 W_{ss}(\hat{\rho}) [g_{ss}(\hat{\rho}, t_f)]^2 d\hat{\rho}, \quad (8)$$

where  $W_q(\hat{\rho})$  and  $W_{ss}(\hat{\rho})$  are positive weight functions and

$$g_{ss}(\hat{\rho}, t) = \frac{\partial U_p}{\partial \hat{\rho}} = -\frac{\partial \Psi}{\partial t} = -2\pi \frac{\partial \psi}{\partial t}, \quad (9)$$

where  $U_p$  is the loop-voltage profile which can be related to the temporal derivative of the poloidal magnetic flux.

# Formulation of Various Constraints

- **Actuator Trajectory Parametrization:** The trajectories of the  $i$ -th control actuator ( $u_i$ ) can be parametrized by a finite number  $n_{p_i}$  of *to-be-determined parameters* ( $x_i$ ) at discrete points in time ( $t_{p_i}$ ), i.e.,

$$t_{p_i} = [t_1, t_2, \dots, t_k, \dots, t_{n_{p_i}} = t_f] \in \mathbb{R}^{n_{p_i}}$$

$$x_i = [u_i^1, u_i^2, \dots, u_i^k, \dots, u_i^{n_{p_i}}] \in \mathbb{R}^{n_{p_i}}$$

- Combining all parameters to represent individual actuator trajectories into the vector  $\tilde{x}$ , where  $\tilde{x} \in \mathbb{R}^{n_p^{tot}}$  and  $n_p^{tot} = \sum_{i=1}^{n_{act}} n_{p_i}$ , the control actuator trajectories can be written compactly as

$$\boxed{u(t) = \Pi(t)\tilde{x}} \quad (10)$$

- **Actuator Constraints:** The actuator magnitude and rate constraints are given by

$$I_p^{min} \leq I_p(t) \leq I_p^{max},$$

$$P_i^{min} \leq P_i(t) \leq P_i^{max}, \quad i = 1, \dots, n_{nbi}$$

$$-I_{p,max}^{d'} \leq dI_p/dt \leq I_{p,max}^{u'}$$

- The above actuator constraints can be written compactly as a matrix inequality as

$$\boxed{A_u^{lim} \tilde{x} \leq b_u^{lim}} \quad (11)$$

# Statement of the Optimization Problem

- The optimization problem is written mathematically as

$$\min_{\tilde{x}} J(t_f) = J(\dot{\theta}(t_f), \theta(t_f)), \quad (12)$$

such that

$$\dot{\theta} = g(\theta, u), \quad (13)$$

$$u(t) = \Pi(t)\tilde{x}, \quad (14)$$

$$A_u^{lim}\tilde{x} \leq b_u^{lim}, \quad (15)$$

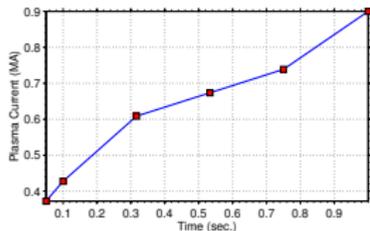
where the **poloidal flux gradient**,  $\theta = d\psi/d\hat{r}$  represents the plasma state,  $u$  represents the actuators, and  $g$  is a nonlinear function representing the plasma dynamics as an additional constraint ((13) is derived from (1)-(6)).

- The optimization problem (12)-(15) can be solved iteratively in MATLAB by using the **Sequential Quadratic Programming (SQP)** method [7].

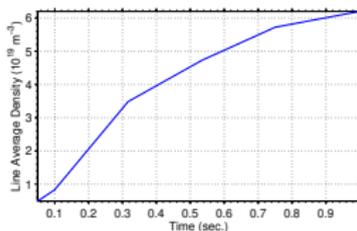
[7] J. Nocedal and S. J. Wright, *Numerical optimization*, (Springer, New York, 2006).

# Feedforward Optimization: Weighting only the $q$ -profile

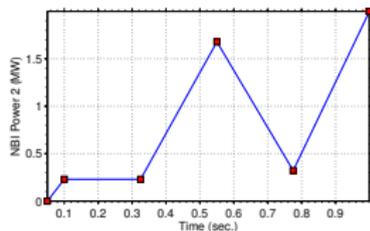
- For this application,  $k_q = 1$  and  $k_{ss} = 0$  in the cost functional.
- **The goal is to hit a target  $q$ -profile at  $t_f = 1$  sec.**



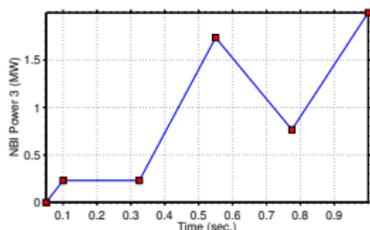
(a) Optimal plasma current  $I_p(t)$



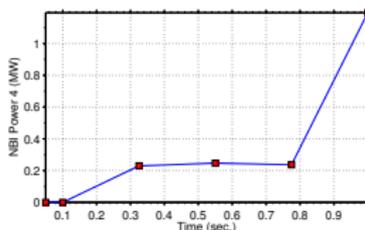
(b) Line Averaged Density  $\bar{n}_e(t)$



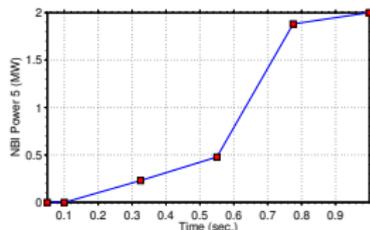
(c) Optimal NBI beam power #2



(d) Optimal NBI beam power #3



(e) Optimal NBI beam power #4

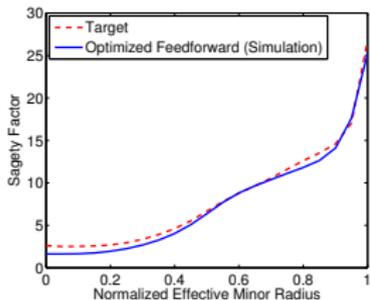


(f) Optimal NBI beam power #5

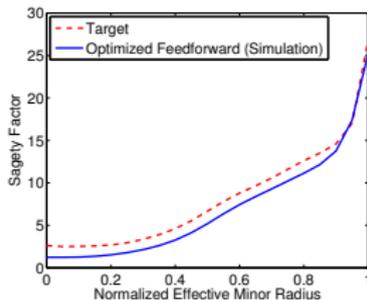
Time evolution of the optimized feedforward actuator trajectories

# Feedforward Optimization: Weighting only the $q$ -profile

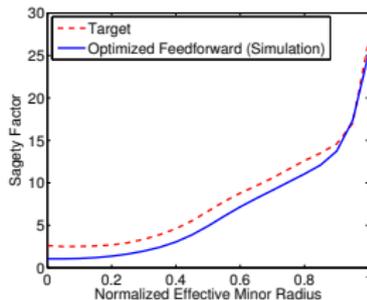
- Comparison of the target and achieved  $q$ -profiles at various times:



(a)  $t = t_f = 1.0$  sec.

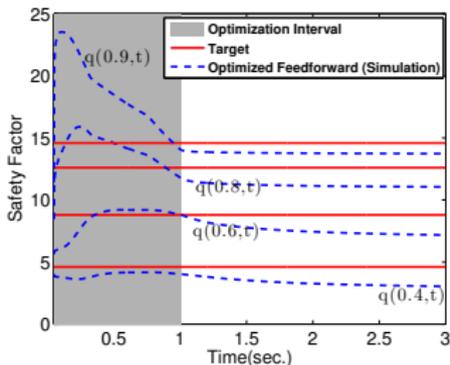


(b)  $t = 2.0$  sec.



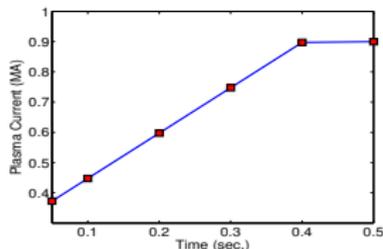
(c)  $t = 3.0$  sec.

- Time evolution of the safety factor at various radial locations:

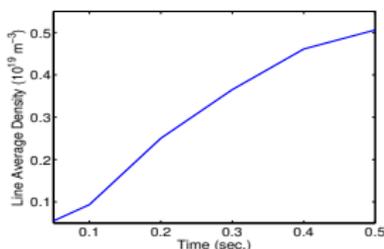


# Feedforward Optimization: Weighting only Steadiness

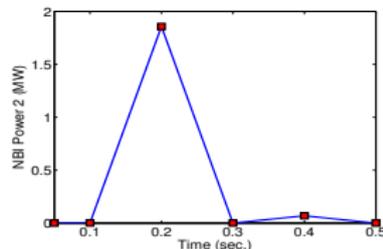
- For this application,  $k_{ss} = 1$  and  $k_q = 0$  in the cost functional.
- **The goal is to maintain a steady  $q$ -profile throughout the simulation.**



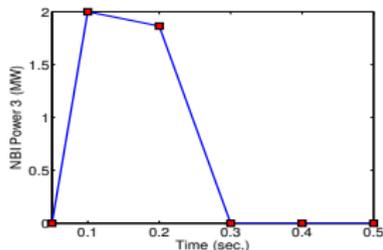
(a) Optimal plasma current  $I_p(t)$



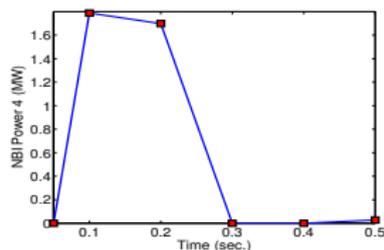
(b) Line Averaged Density



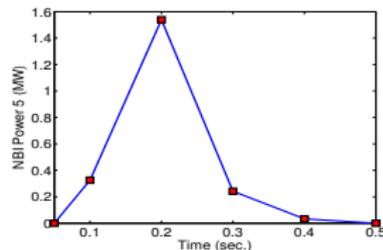
(c) Optimal NBI beam power #2



(d) Optimal NBI beam power #3



(e) Optimal NBI beam power #4

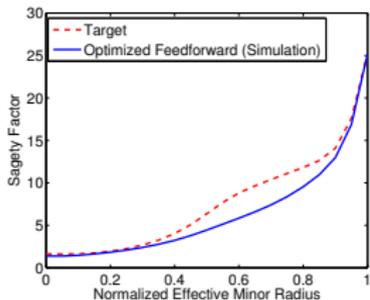


(f) Optimal NBI beam power #5

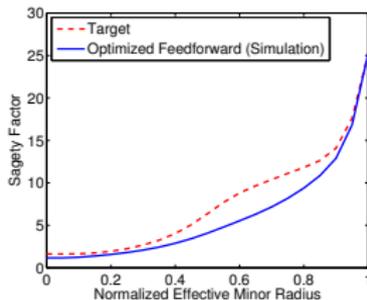
Time evolution of the optimized feedforward actuator trajectories

# Feedforward Optimization: Weighting only Steadiness

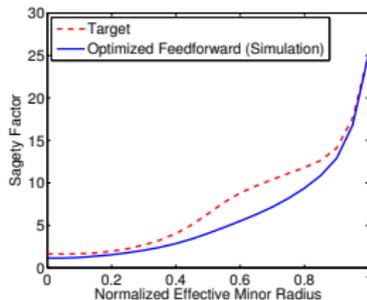
- Comparison of the target and achieved  $q$ -profiles at various times:



(a)  $t = t_f = 0.5$  sec.

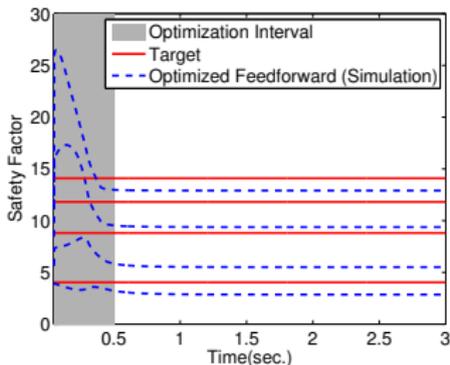


(b)  $t = 1.0$  sec.



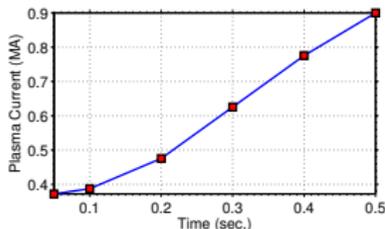
(c)  $t = 2.0$  sec.

- Time evolution of the safety factor at various radial locations:

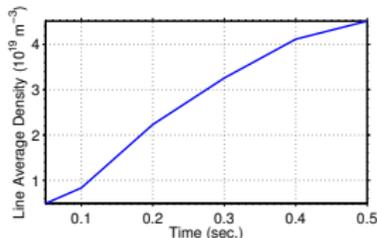


# Feedforward Optimization: Weighting $q$ + Steadiness

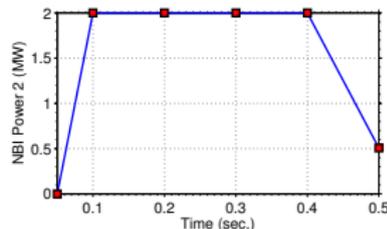
- For this application,  $k_{ss} = 1$  and  $k_q = 1$  in the cost functional.
- **The goal is to hit a target  $q$ -profile at  $t = 0.5$  sec. and maintain it throughout a 3 sec. simulation.**



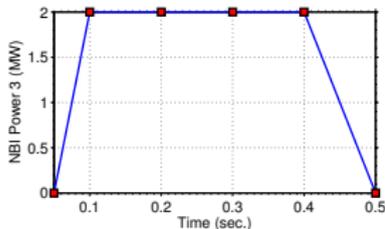
(a) Optimal plasma current  $I_p(t)$



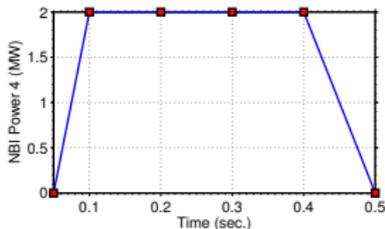
(b) Line Averaged Density



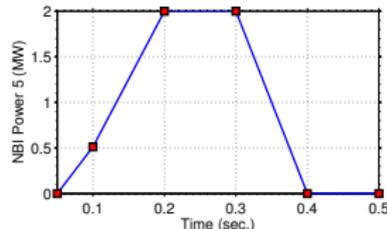
(c) Optimal NBI beam power #2



(d) Optimal NBI beam power #3



(e) Optimal NBI beam power #4

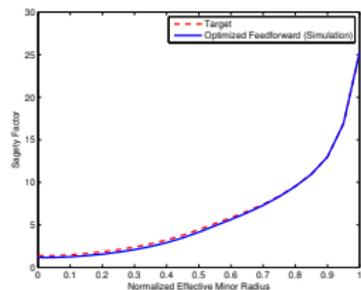


(f) Optimal NBI beam power #5

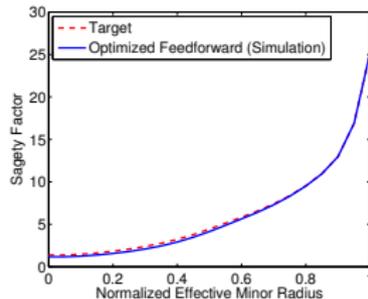
Time evolution of the optimized feedforward actuator trajectories

# Feedforward Optimization: Weighting $q$ + Steadiness

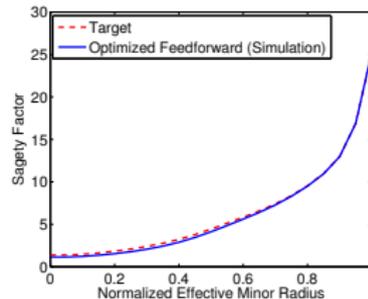
- Comparison of the target and achieved  $q$ -profiles at various times:



(a)  $t = t_f = 0.5$  sec.

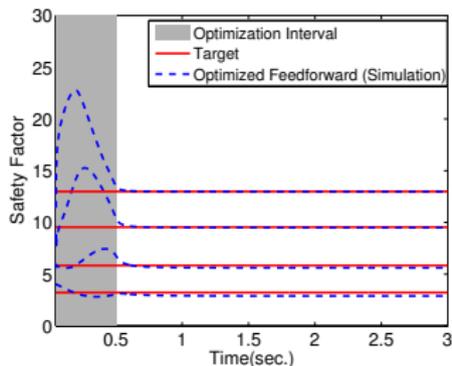


(b)  $t = 1.0$  sec.



(c)  $t = 2.0$  sec.

- Time evolution of the safety factor at various radial locations:



# Optimal Feedback Control of the Current Density Profile

- **Linear-Quadratic-Integral (LQI)** Optimal feedback controller has been designed in **MATLAB** based on the **FPD, control-oriented model**.
- The **effectiveness** of the designed **controller** is **first tested in MATLAB** by simulating the nonlinear **magnetic diffusion equation (1)**.
- Early results on control design and numerical testing have been presented in **[8], [9]**.
- The **proposed feedback controller is now implemented in TRANSP** for performance assessment before experimental testing in NSTX-U.
- Recently developed **Expert routine [10]** provides a **framework** to perform **closed-loop predictive simulations** within the **TRANSP source code**.

[8] ILHAN, Z. O. et al., 56<sup>th</sup> Annual Meeting of the APS DPP (2014)

[9] ILHAN, Z. O. et al., IEEE Multi-Conference on Systems and Control (2015)

[10] BOYER, M. D., ANDRE, R., GATES, D. A., et al., Nuclear Fusion **55 (2015)**

# Closed-Loop Control Simulation Study in TRANSP

- The control objective is to track a target state trajectory  $\iota_r(\rho, t)$  with *minimum control effort*.
- The target state trajectory  $\iota_r(\rho, t)$  is generated through an open-loop TRANSP simulation with the following constant reference inputs.

$\mathbf{n}_e(\text{m}^{-3})$	$5.0 \times 10^{19}$	$\mathbf{P}_4(\text{W})$	$0.8 \times 10^6$
$\mathbf{P}_1(\text{W})$	$0.2 \times 10^6$	$\mathbf{P}_5(\text{W})$	$1.0 \times 10^6$
$\mathbf{P}_2(\text{W})$	$0.4 \times 10^6$	$\mathbf{P}_6(\text{W})$	$1.2 \times 10^6$
$\mathbf{P}_3(\text{W})$	$0.6 \times 10^6$	$\mathbf{I}_p(\text{A})$	$0.7 \times 10^6$

- Starting from the first second of the simulation, the controller is tested against *perturbed initial conditions* and *constant input disturbances*, i.e.,

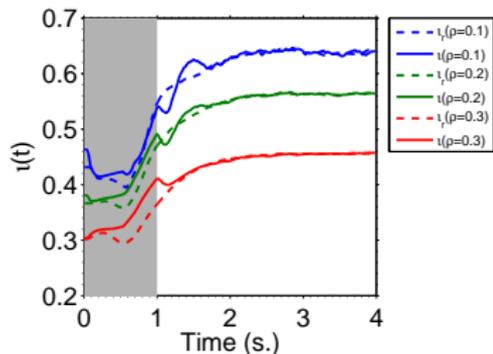
$$u(t) = \begin{cases} u_r + u_d, & t \leq 1 \text{ s.} \\ u_r + u_d + \Delta u(t), & t > 1 \text{ s.} \end{cases}$$

where  $u_r$  represents the constant reference inputs,  $u_d$  stands for the constant disturbance inputs (15% for  $I_p$ , 10% for  $P_1$ ,  $P_3$ ,  $P_5$  and  $P_6$ ), and  $\Delta u(t)$  is the output of the feedback controller.

# CASE 1: Actuation with $I_p$ and Neutral Beams

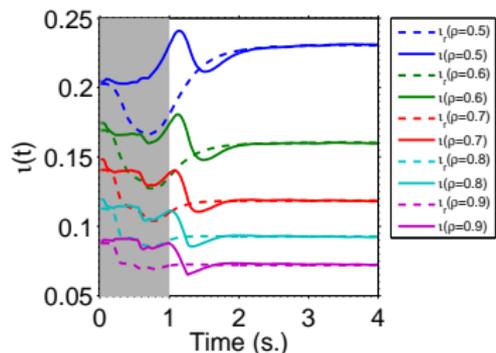
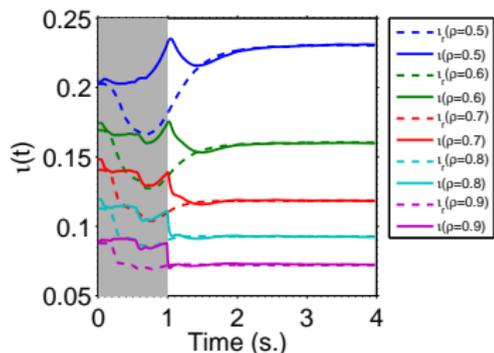
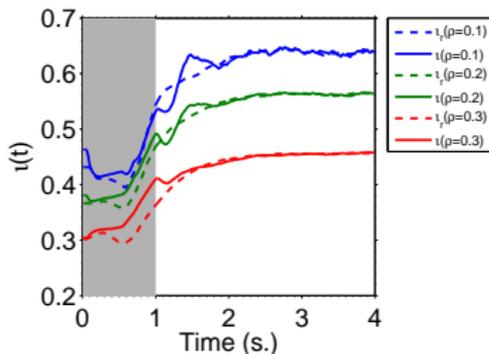
## CASE 1A

(without  $I_p$  rate saturation)



## CASE 1B

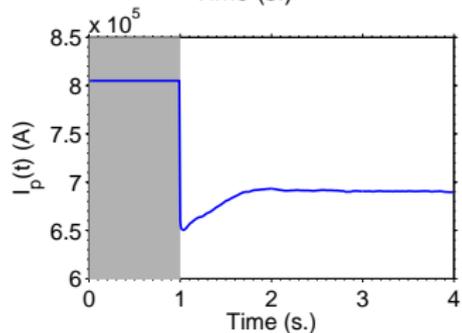
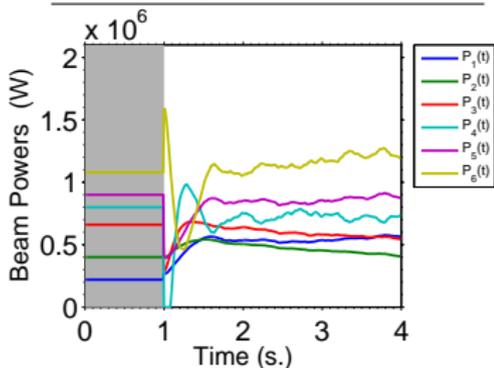
(with  $I_p$  rate saturation)



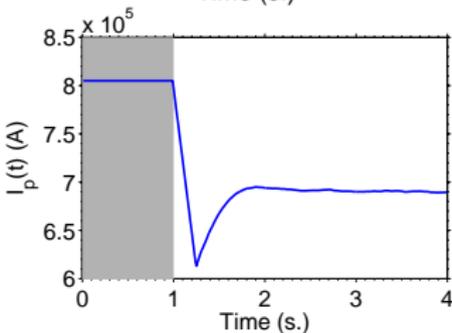
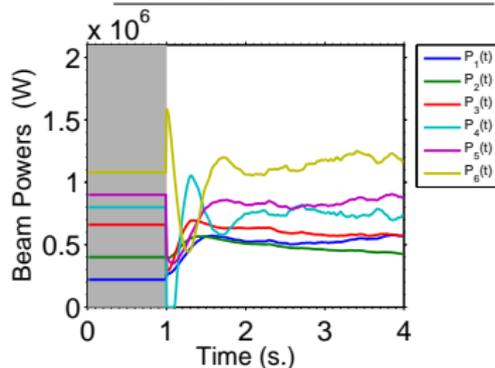
Figures (left & right): Time evolution of the optimal outputs (solid) with their respective targets (dashed). Controller is off in the grey region.

# CASE 1: Actuation with $I_p$ and Neutral Beams

**CASE 1A**  
(without  $I_p$  rate saturation)



**CASE 1B**  
(with  $I_p$  rate saturation)

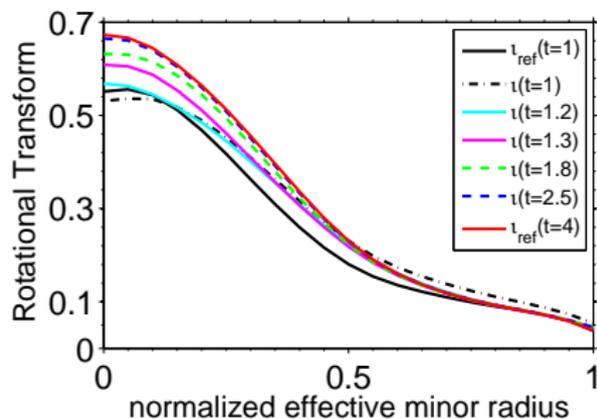


Figures (upper left & right): Time evolution of the optimal beam powers.  
Figures (lower left & right): Time evolution of the optimal plasma current.

# CASE 1: Actuation with $I_p$ and Neutral Beams

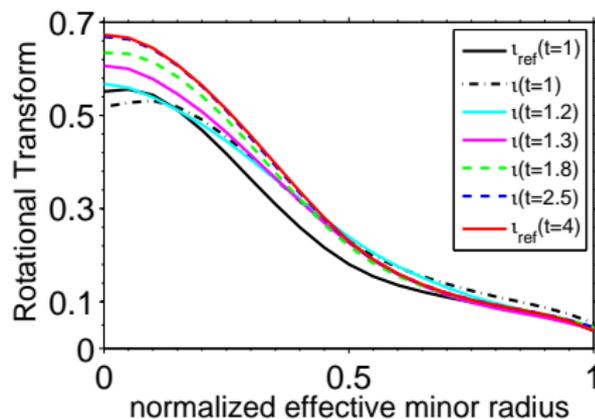
## CASE 1A

(without  $I_p$  rate saturation)



## CASE 1B

(with  $I_p$  rate saturation)

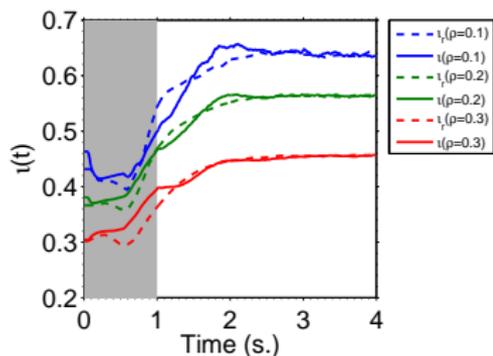


Figures (left & right): Time evolution of the rotational transform ( $l$ -profile).

# CASE 2: Actuation with $n_e$ , $I_p$ and Neutral Beams

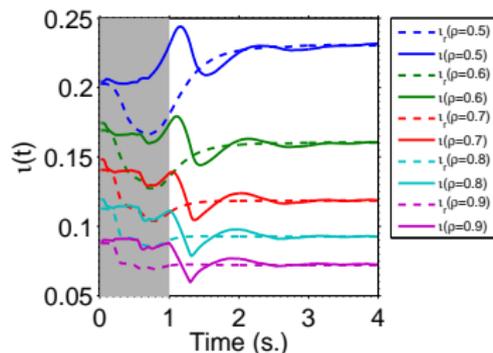
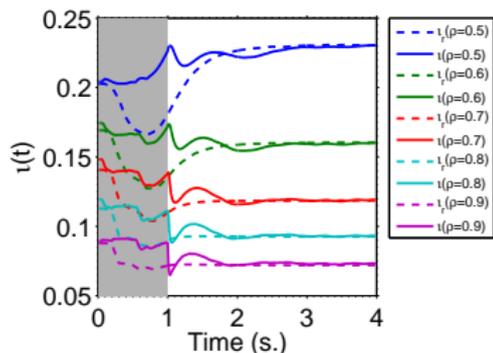
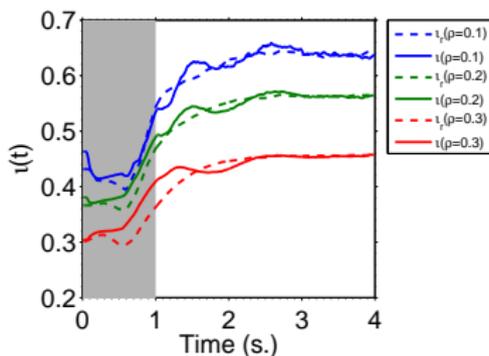
## CASE 2A

(without  $I_p$  &  $n_e$  rate saturation)



## CASE 2B

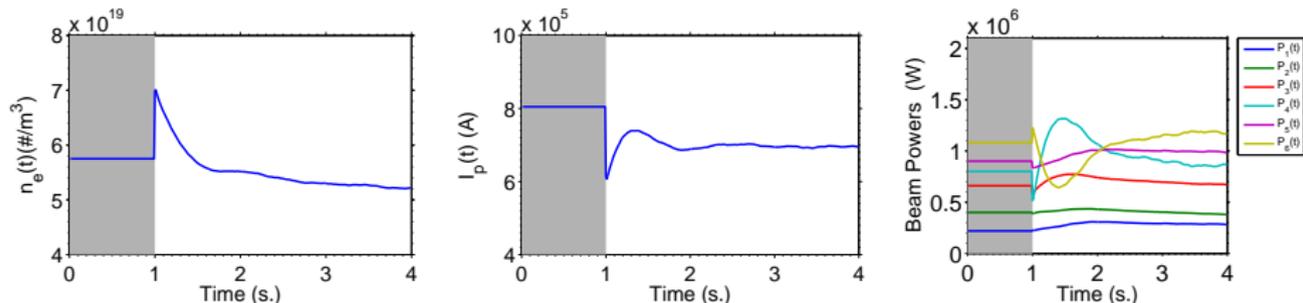
(with  $I_p$  &  $n_e$  rate saturation)



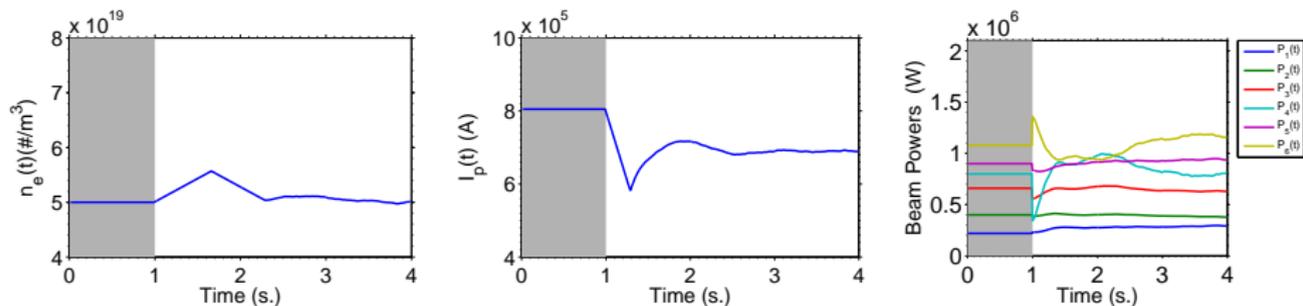
Figures (left & right): Time evolution of the optimal outputs (solid) with their respective targets (dashed). Controller is off in the grey region.

# CASE 2: Actuation with $n_e$ , $I_p$ and Neutral Beams

## ● CASE 2A (without $I_p$ & $n_e$ rate saturation)



## ● CASE 2B (with $I_p$ & $n_e$ rate saturation)

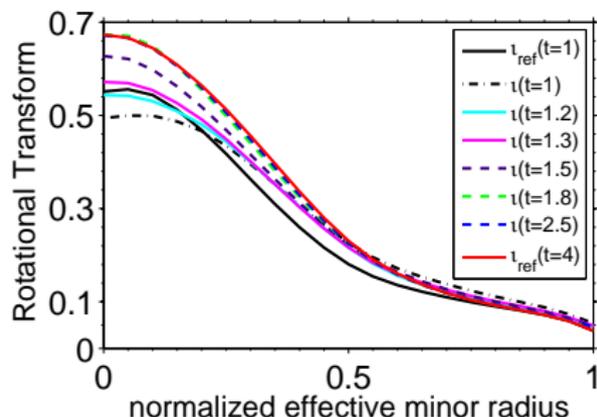


Figures: (left) Time evolution of the optimal line-averaged electron density, (center) time evolution of the optimal plasma current, and (right) time evolution of the optimal beam powers.

# CASE 2: Actuation with $n_e$ , $I_p$ and Neutral Beams

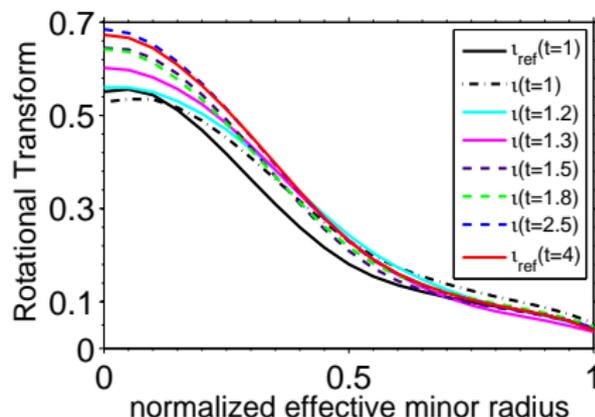
## CASE 2A

(without  $I_p$  &  $n_e$  rate saturation)



## CASE 2B

(with  $I_p$  &  $n_e$  rate saturation)



Figures (left & right): Time evolution of the rotational transform ( $l$ -profile).

## CASE 3: Control Against Changing Confinement Factor

- In predictive TRANSP simulations,  $n_e$  and  $T_e$  profile evolutions are *not modeled by first-principles calculations*. [11]
- A reference  $n_e$  profile is specified based on an experimental profile measured on NSTX and then scaled to achieve a particular Greenwald fraction,  $f_{GW}$ .
- Similarly,  $T_e$  profile is also taken from an experiment and scaled to achieve a particular global confinement time [12]

$$\tau_{ST} = H_{ST} 0.1178 I_p^{0.57} B_T^{1.08} n_e^{0.44} P_{Loss,th}^{-0.73}$$

- When performing closed-loop simulations in TRANSP, the simulation **must be constrained to follow a specific confinement level all the time** although the actuators are varied based on the calculations of the feedback controller  $\Rightarrow$  This is achieved by **manipulating** the confinement factor  $H_{ST}$  through a **user-defined waveform**. [13]
- However, the  $H_{ST}$  factor **can deviate** from the user-supplied waveform in the NSTX-U experiments  $\Rightarrow$  **creating additional source of disturbance**.

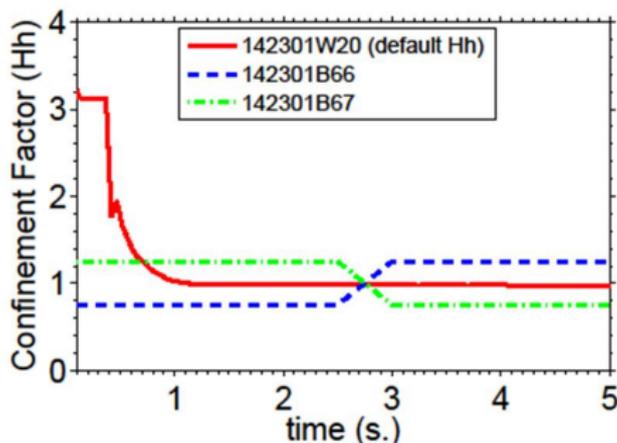
[11] GERHARDT, S. P., ANDRE, R., and MENARD, J. E., Nuclear Fusion **52** (2012).

[12] KAYE, S. et al., Nuclear Fusion **46** (2006).

[13] BOYER, M. D., ANDRE, R., GATES, D. A., et al., Nuclear Fusion **55** (2015).

## CASE 3: Control Against Changing Confinement Factor

- Two closed-loop TRANSP simulations are carried out to verify disturbance rejection against changing confinement factors:
  - Run 142301B66 has a step increase in the  $H_{ST}$  from 0.75 to 1.25.
  - Run 142301B67 has a step decrease in the  $H_{ST}$  from 1.25 to 0.75.
- Note that the target profile corresponds to the open-loop run 142301W20, which has  $H_{ST} \approx 1$  when  $t \in [1 \text{ } 5]$  s., during which the controller is on.
- Only  $I_p$  and neutral beams are used as actuators without considering rate saturations.

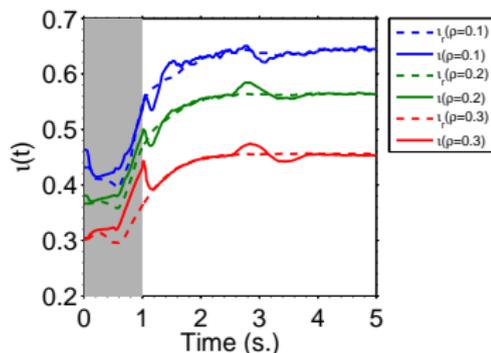


Time evolution of the confinement factors.

# CASE 3: Control Against Changing Confinement Factor

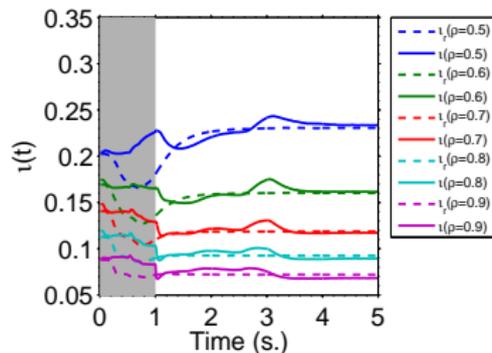
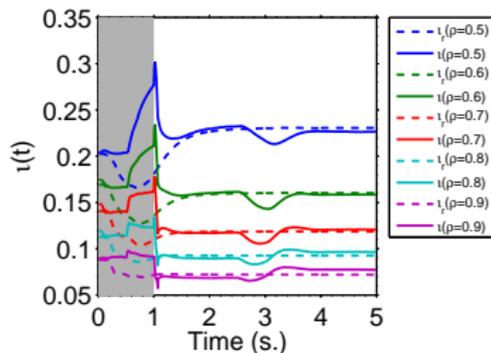
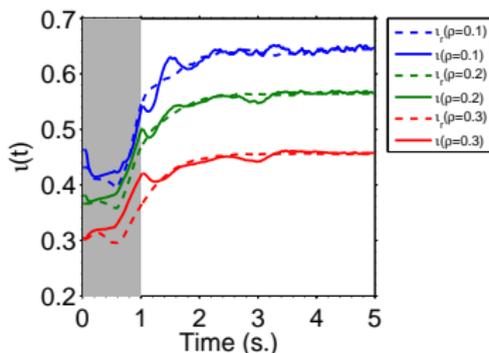
## CASE 3A

(increasing  $H_{ST}$  - 142301B66)



## CASE 3B

(decreasing  $H_{ST}$  - 142301B67)

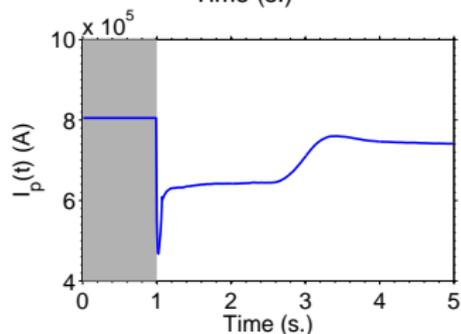
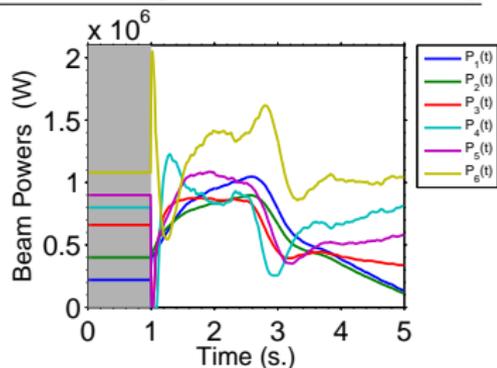


Figures (left & right): Time evolution of the optimal outputs (solid) with their respective targets (dashed). Controller is off in the grey region.

# CASE 3: Control Against Changing Confinement Factor

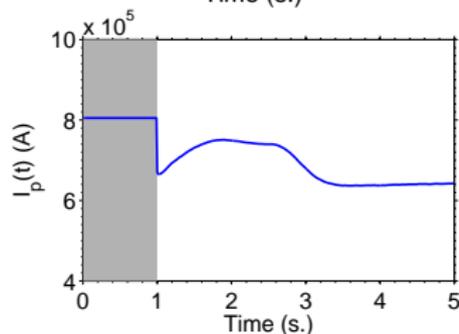
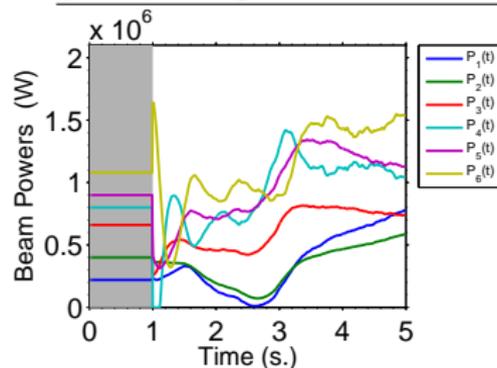
## CASE 3A

(increasing  $H_{ST}$  - 142301B66)



## CASE 3B

(decreasing  $H_{ST}$  - 142301B67)

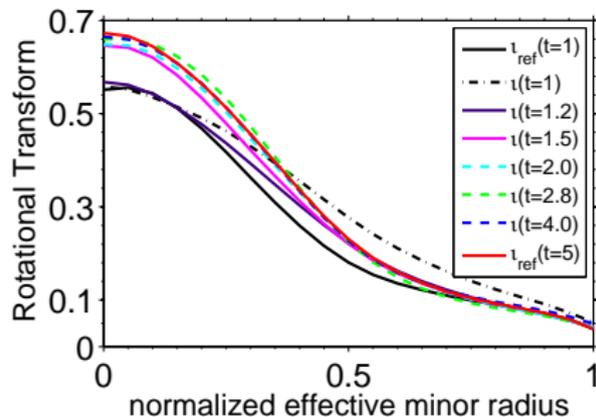


Figures (upper left & right): Time evolution of the optimal beam powers.  
Figures (lower left & right): Time evolution of the optimal plasma current.

# CASE 3: Control Against Changing Confinement Factor

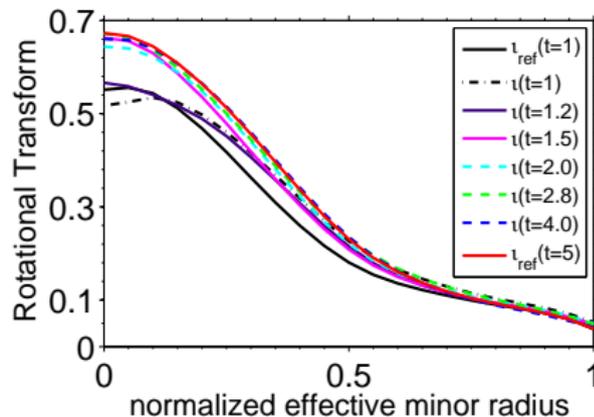
## CASE 3A

(increasing  $H_{ST}$  - 142301B66)



## CASE 3B

(decreasing  $H_{ST}$  - 142301B67)



Figures (left & right): Time evolution of the rotational transform ( $l$ -profile).

# Conclusion and Future Work

- A nonlinear, control-oriented, physics-based model has been proposed to describe the evolution of the poloidal magnetic flux profile, which can be related to the  $q$ -profile ( $\iota$ -profile)  $\Rightarrow$  the current density profile.
- Using this first-principles-driven (FPD), control-oriented model, a **two-component control design** approach has been proposed for the regulation of the current density profile:
  - 1 **A feedforward trajectory optimizer** (controller) to compute offline actuator requests to achieve specific plasma scenarios.
  - 2 **A feedback control algorithm** to track a desired current density profile while adding robustness against model uncertainties and disturbances to the overall current profile control scheme.
- **The performance of the feedback controller has been validated in TRANSP simulations through the recently developed Expert routine**, which provides a framework to perform closed-loop predictive simulations within the TRANSP source code.
- **The immediate next step** is to test the feedforward actuator trajectory optimizer in TRANSP and then in the actual NSTX-U machine.
- **A longer-term next step** is the implementation of the feedback controller in the NSTX-U PCS with the ultimate goal of experimental testing.